

Fatigue and lifetime prediction strategies for polymers

Tom Engels (DSM Research, TU/e), Leon Govaert (TU/e)



amorphous vs. semi-crystalline polymers



polymer glasses



e.g. polycarbonate



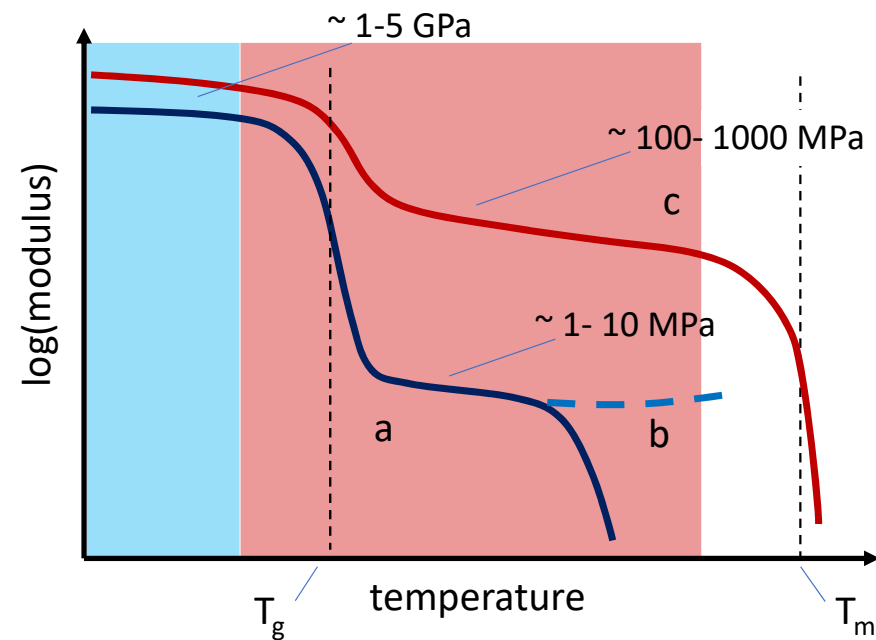
semi-crystalline polymers



e.g. polyamide-6

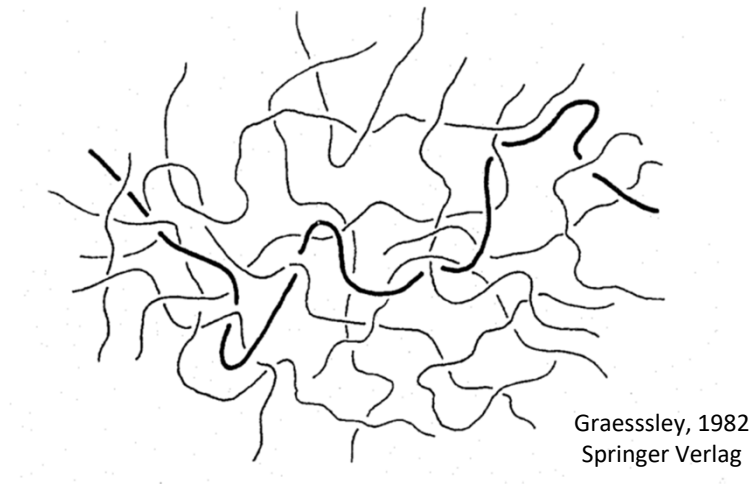
application ranges

- a. amorphous polymer
(e.g. PS, PC, PMMA)
- b. crosslinked amorphous polymer
(thermosets)
- c. semi-crystalline polymer
(e.g. PE, PP, PA, PET)



- amorphous polymers are used (well) below their T_g
- semi-crystalline polymers are used (well) below their T_m
typically crystallinities are optimized to achieve maximum modulus

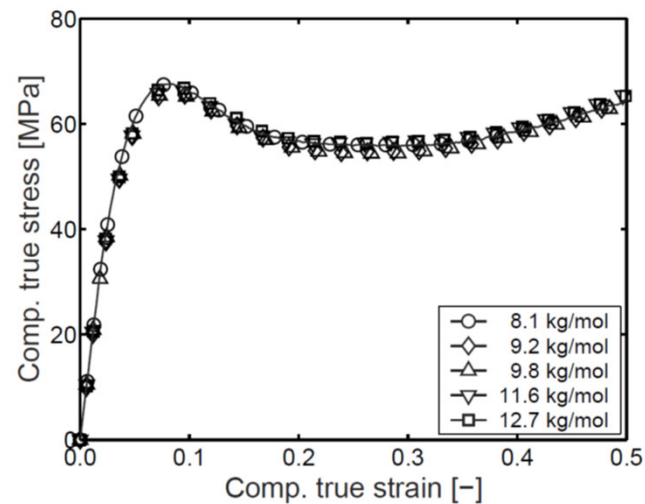
polymer glasses



no long range order present in the material;
entanglements (M_e) and chain-ends (M_n) dominate the performance

polymer glasses

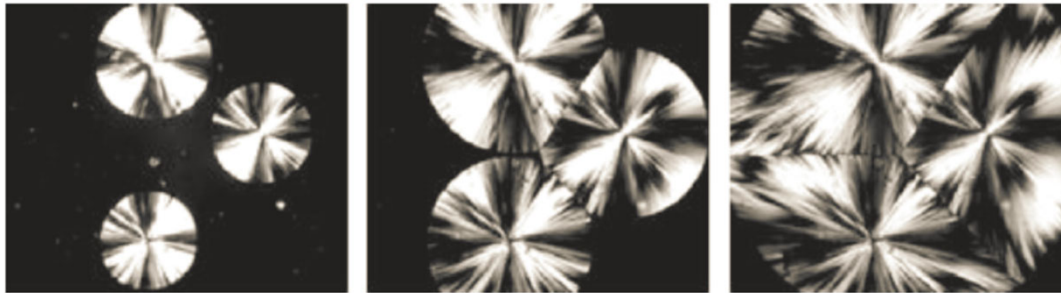
intrinsic behavior



- intrinsic deformation behavior independent of molecular weight; strain hardening $\sim M_e$
- yield stress level depends on cooling rate, but is independent of molecular weight

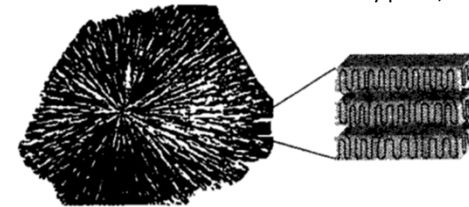
semi-crystalline polymers

Strobl, The Physics of Polymers, Springer, 2007

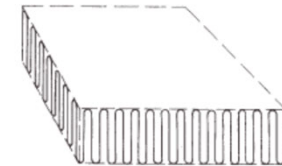


growing spherulites observed during the crystallization of PLLA in an optical microscope (polarized light)

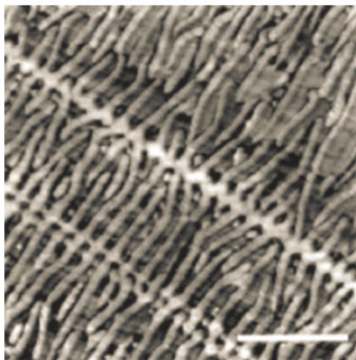
Rubinstein & Colby, Polymer Physics, Oxford University press, 2003



schematic representation of chain folding in polymer crystals



Keller, Rep. Progr. Phys.,
31, 623 (1969)



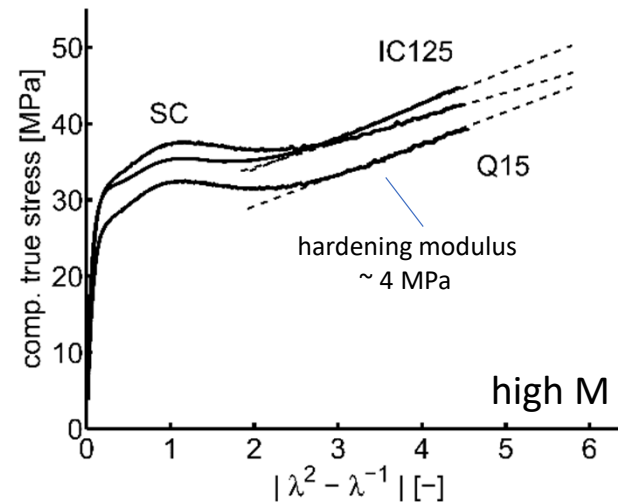
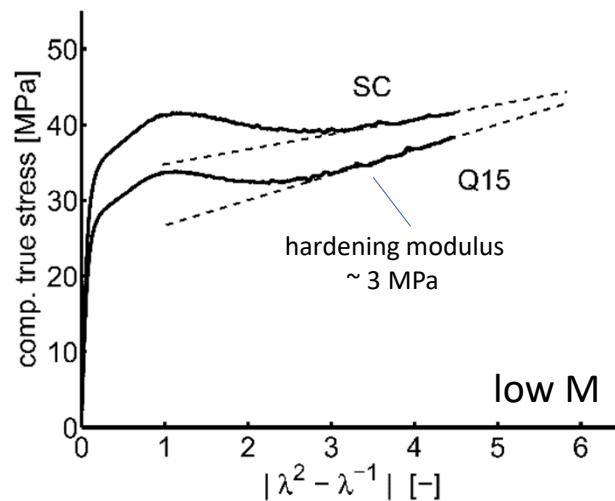
AFM image depicting a PE shish kebab structure (*scale bar*: 300 nm).

Strobl, The Physics of Polymers,
Springer, 2007

crystal morphologies are very rich and dominate the mechanical performance; M_w does influence performance via morphology

semi-crystalline polymers

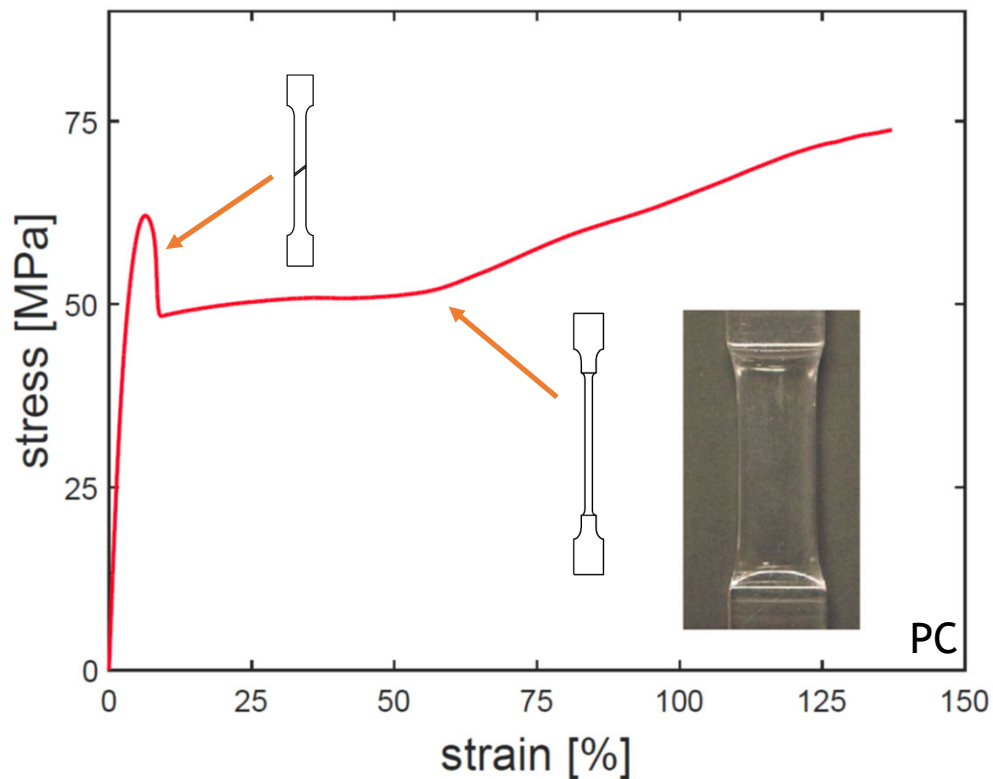
intrinsic behavior



SC = slow cooled
Q15 = quenched at 15°C
IC125 = isothermally crystallized @125°C

- intrinsic deformation behavior depends strongly on molecular weight; strain hardening $\sim M_e$, $\sim M_w$ and crystallization conditions
- yield stress level depends on cooling rate via rate of crystallization ($\sim M_w$)

tensile test / what do we measure ?



engineering stress:

$$\sigma = \frac{F}{A_0}$$

engineering strain:

$$\varepsilon = \frac{\Delta L}{L_0} \cdot 100\%$$

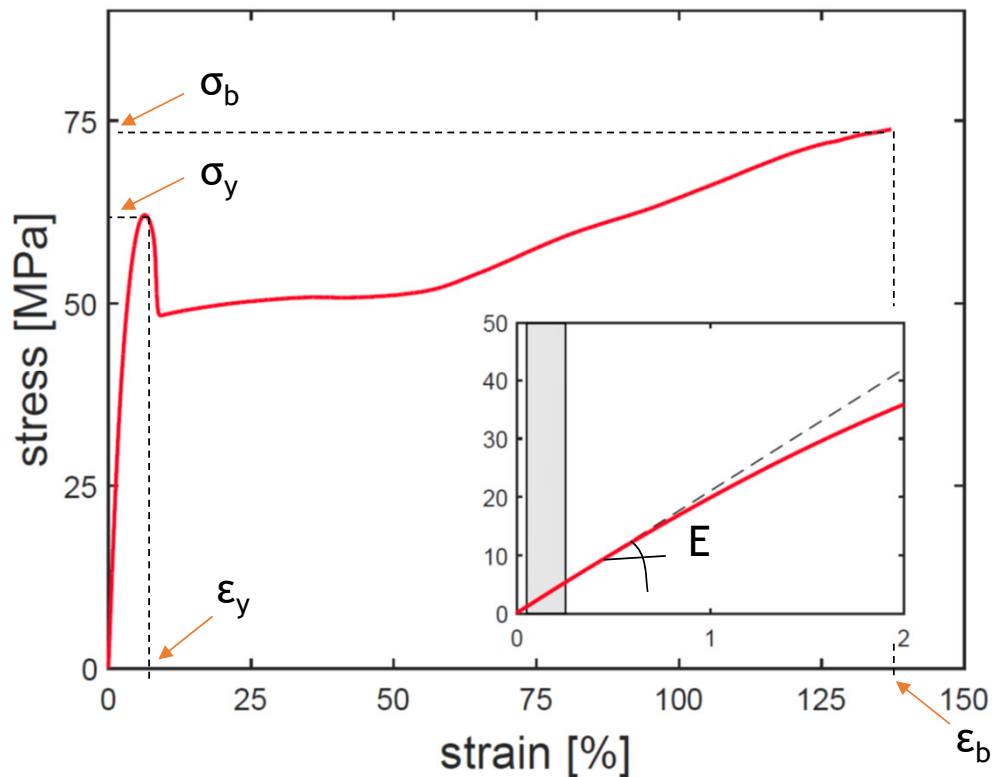
measured *during* test

$\Delta L, F$

A_0, L_0
measured
before test

ΔL = elongation [mm]
 F = force [N]
 A_0 = cross-sectional area [mm²]
 L_0 = initial specimen length [mm]

tensile test / what do we measure ?

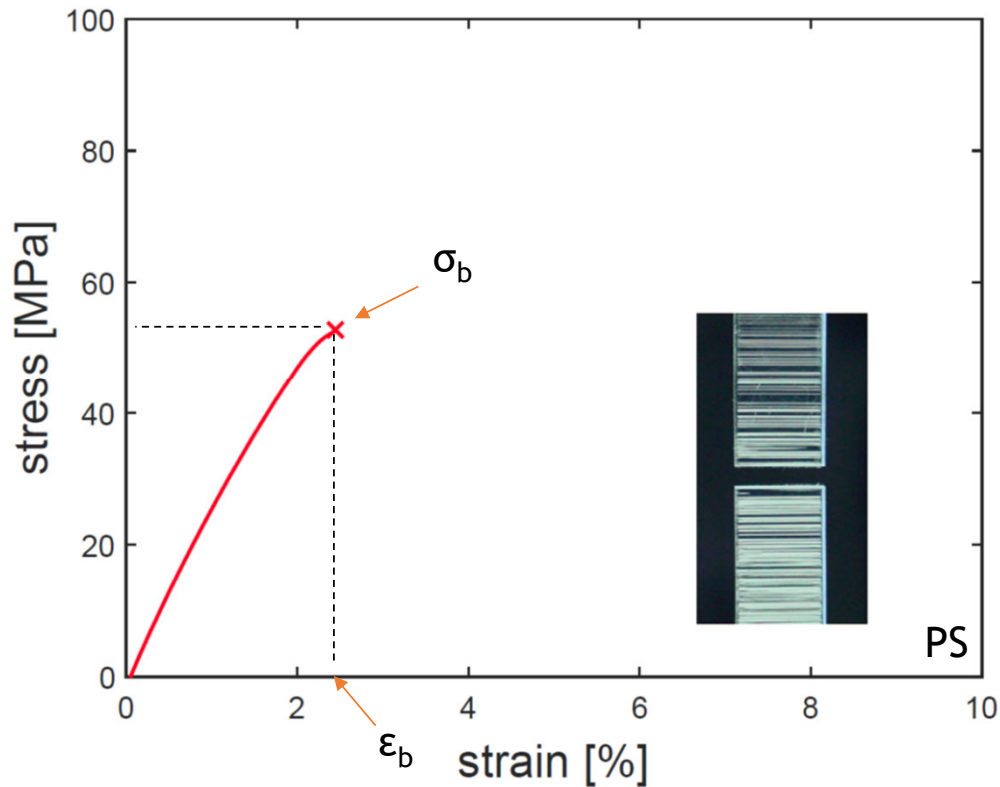


Which material parameters are determined?

- Young's modulus, E
(secant modulus, typically $0.05\% < \epsilon < 0.25\%$)
- yield stress, σ_y
yield strain, ϵ_y
- strength at break, σ_b
strain at break, ϵ_b
(elongation at break, eab)

Example of ductile behavior

tensile test / what do we measure ?

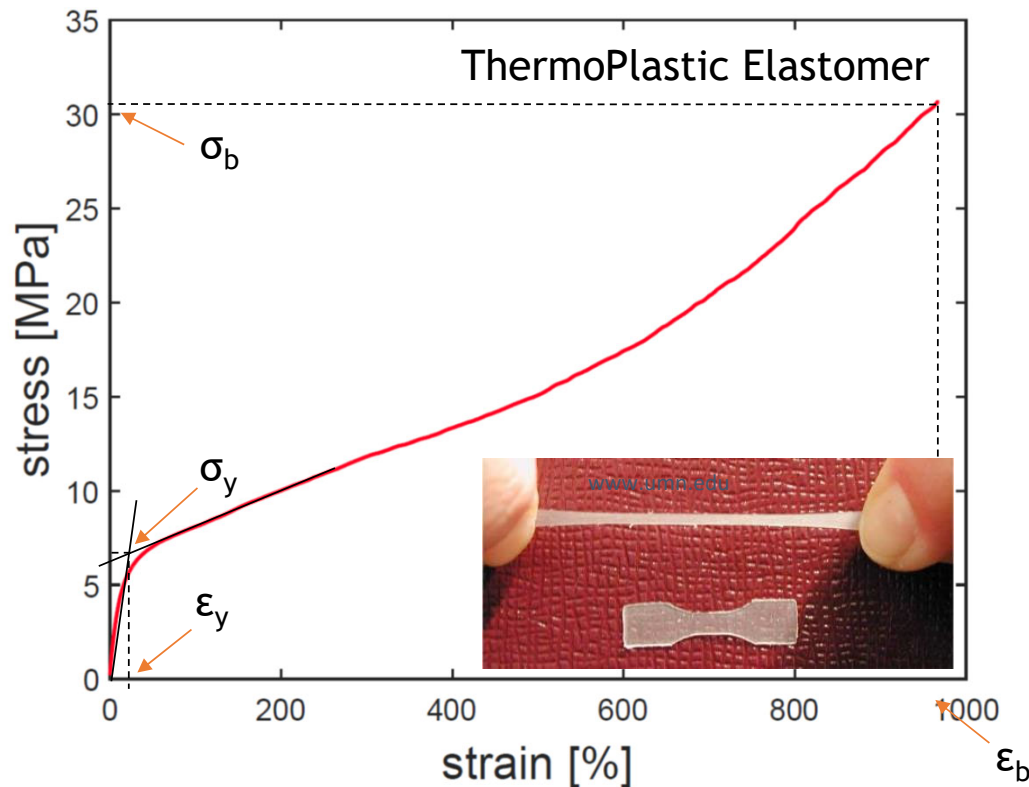


Which material parameters are determined?

- Young's modulus, E
(secant modulus, typically $0.05\% < \epsilon < 0.25\%$)
- yield stress, σ_y
yield strain, ϵ_y
- strength at break, σ_b
strain at break, ϵ_b
(elongation at break, eab)

Example of brittle behavior

tensile test / what do we measure ?

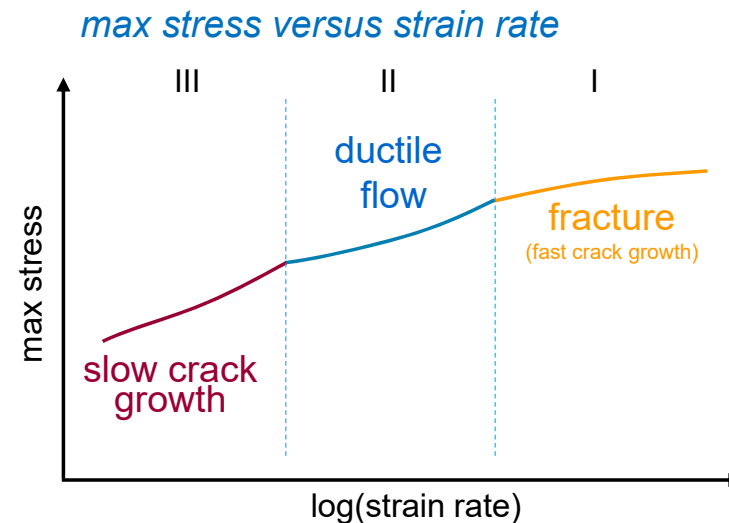
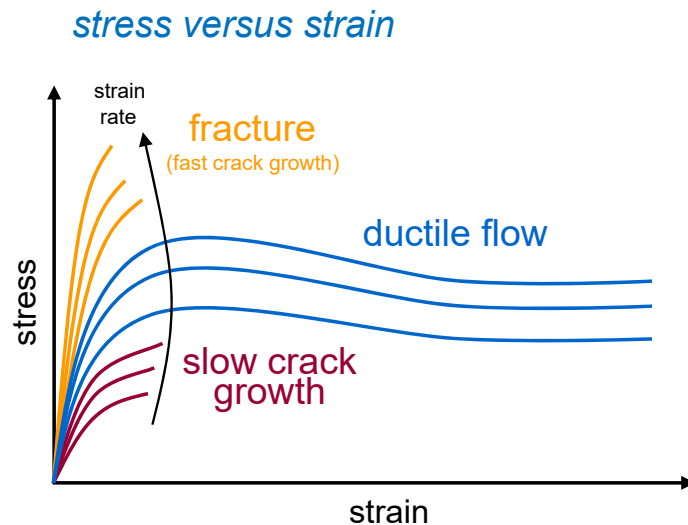


Which material parameters are determined?

- Young's modulus, E
(secant modulus, typically $0.05\% < \epsilon < 0.25\%$)
- yield stress, σ_y
yield strain, ϵ_y
- strength at break, σ_b
strain at break, ϵ_b
(elongation at break, eab)

Elastomeric (like) materials typically do not show a distinct yield point / homogeneous deformation

properties depend on testing conditions



Failure of polymers over broad ranges of strain rate can be assigned to three failure modes:

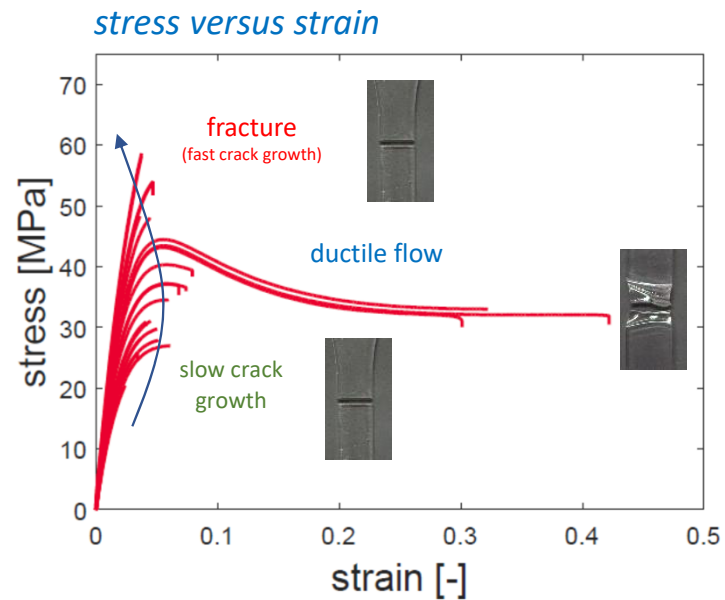
- I. high strain rates / short times:
- II. intermediate strain rates:
- III. low strain rates / long times:

fracture (fast crack growth)

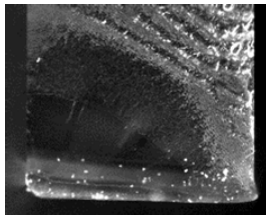
ductile flow

slow crack growth

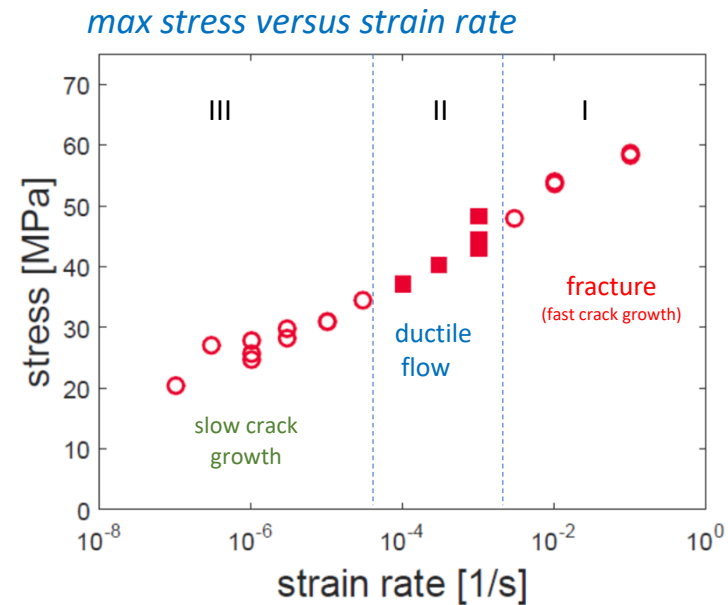
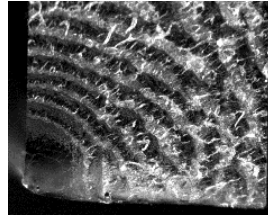
properties depend on testing conditions



Slow Crack Growth



Fracture

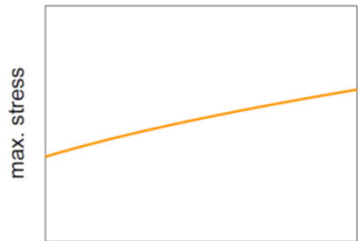


example on commercial PMMA material:

- injection molded tensile bars
- measured @60°C

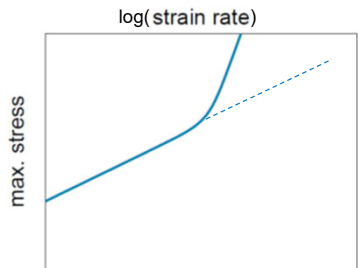
properties depend on testing conditions

typical response for:



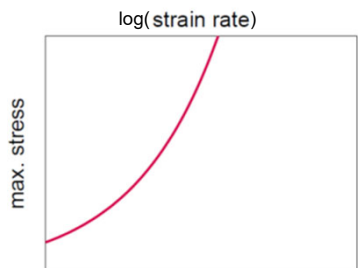
Fracture (Griffith's law)

main parameters are energy release rate ($=f(\text{mol. weight})$) and initial flaw size



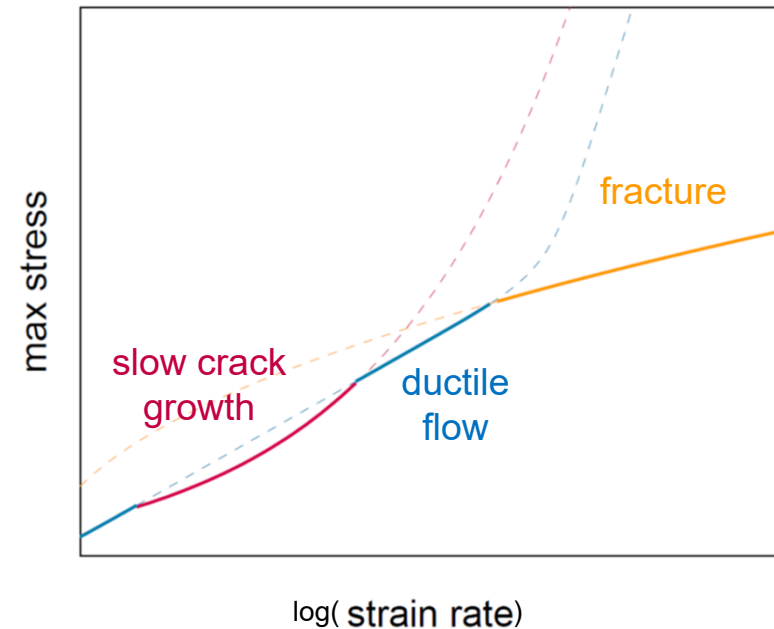
ductile flow (Eyring's flow rule)

main influence is the thermal history ($\neq f(\text{mol. weight})$)



slow crack growth (Paris' law)

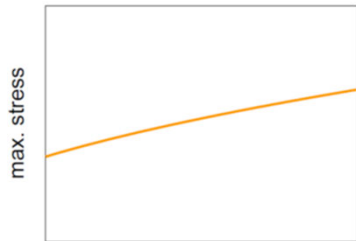
main parameters are the crack growth kinetics ($=f(\text{mol. weight})$) and initial flaw size



failure mechanisms observed in experiment will depend strongly on temperature and rate

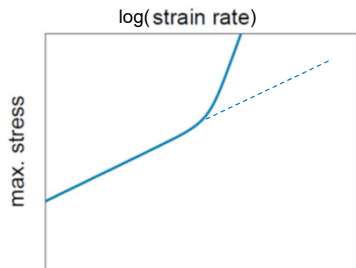
properties depend on testing conditions

typical response for:



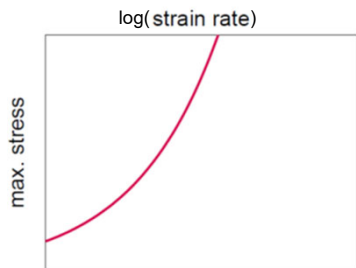
Fracture (Griffith's law)

main parameters are energy release rate ($=f(\text{mol. weight})$) and initial flaw size



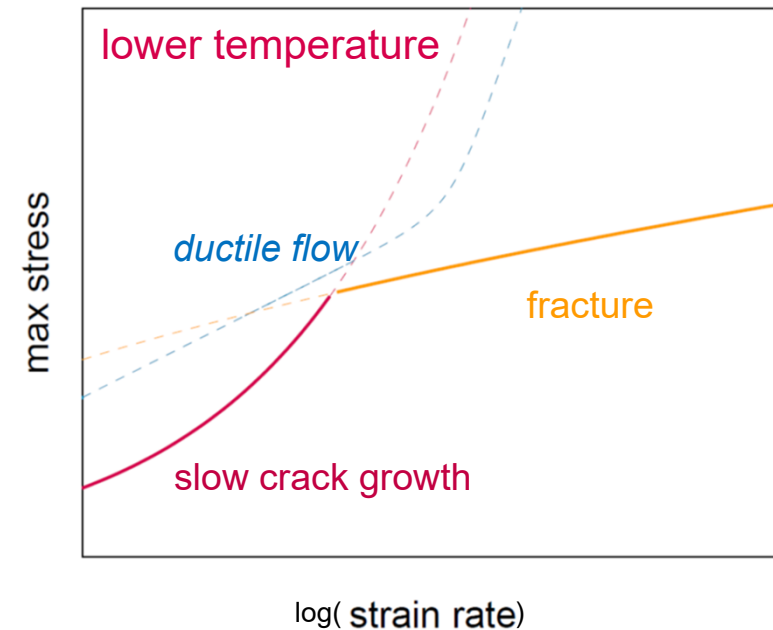
ductile flow (Eyring's flow rule)

main influence is the thermal history ($\neq f(\text{mol. weight})$)



slow crack growth (Paris' law)

main parameters are the crack growth kinetics ($=f(\text{mol. weight})$) and initial flaw size

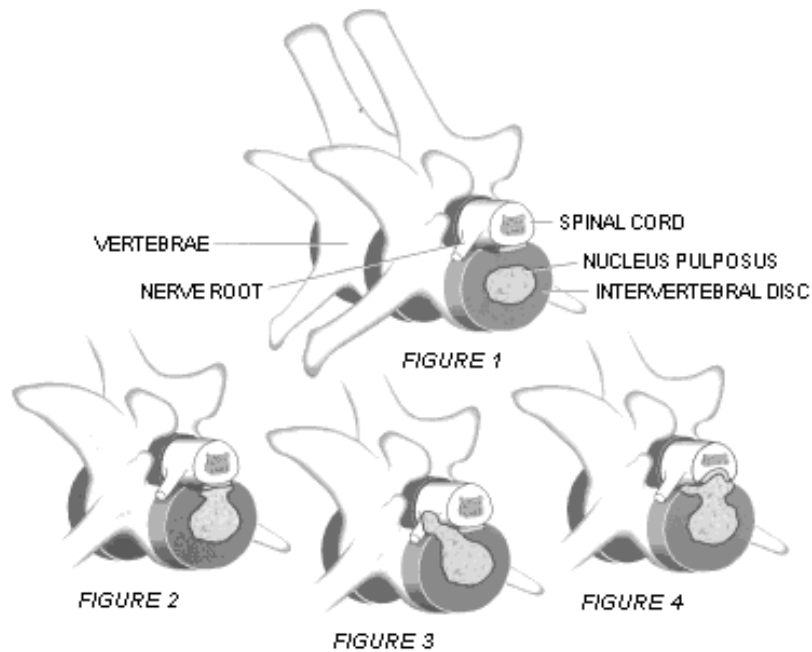


failure mechanisms observed in experiment will depend strongly on temperature and rate

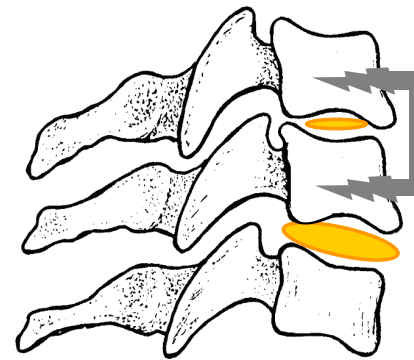
Fatigue and lifetime prediction strategies for polymers

an illustrative example

intervertebral disc degeneration



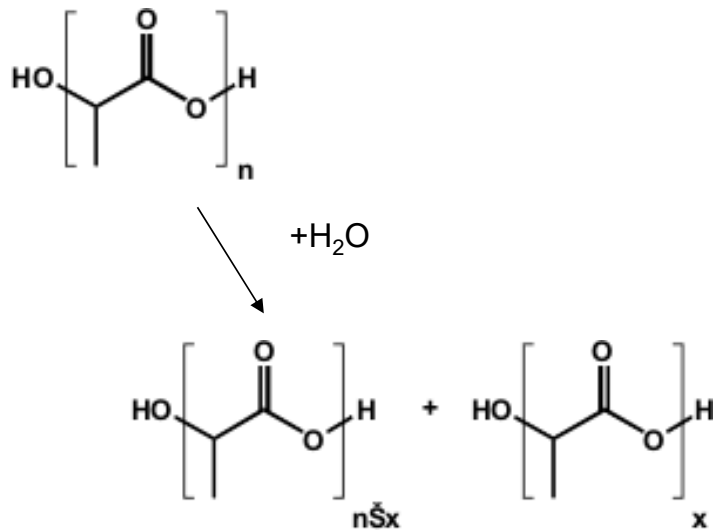
acute case: hernia



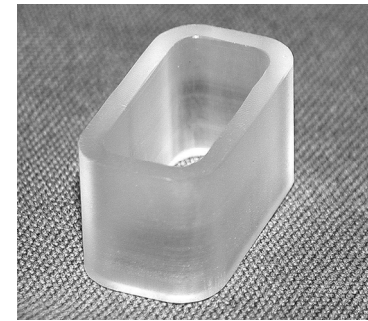
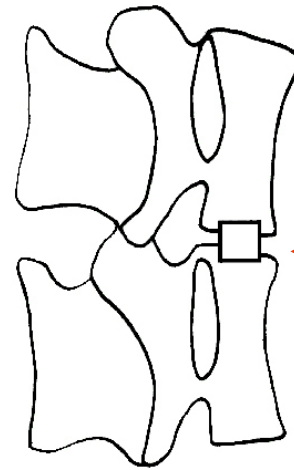
fixation of adjacent
vertebral bodies

interbody fusion using bioresorbable spinal cages

material degrades over time

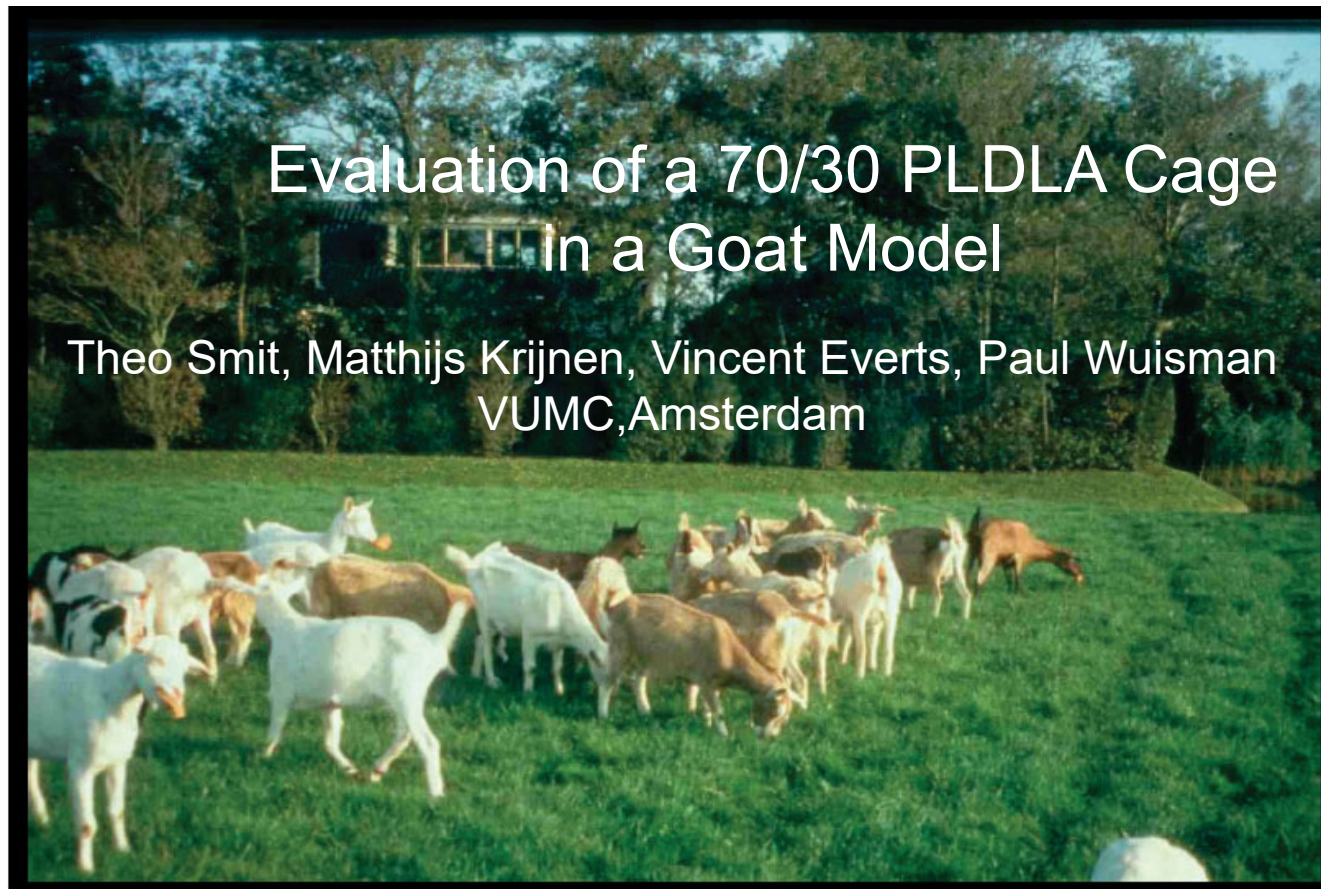


→ loss in molecular weight

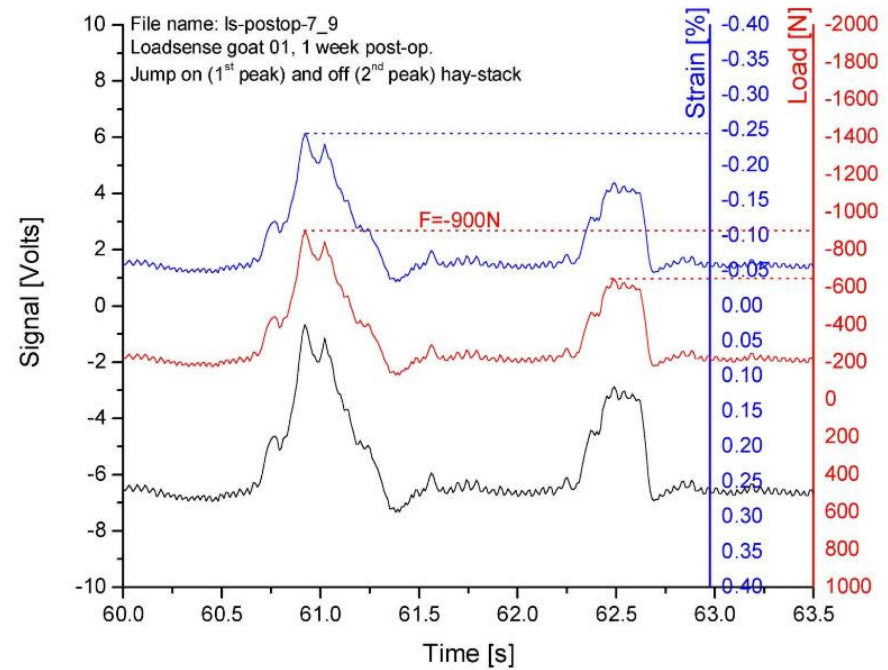


→ loss in strength !!

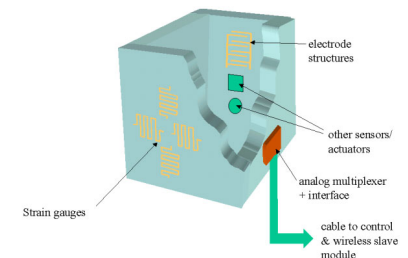
an illustrative example



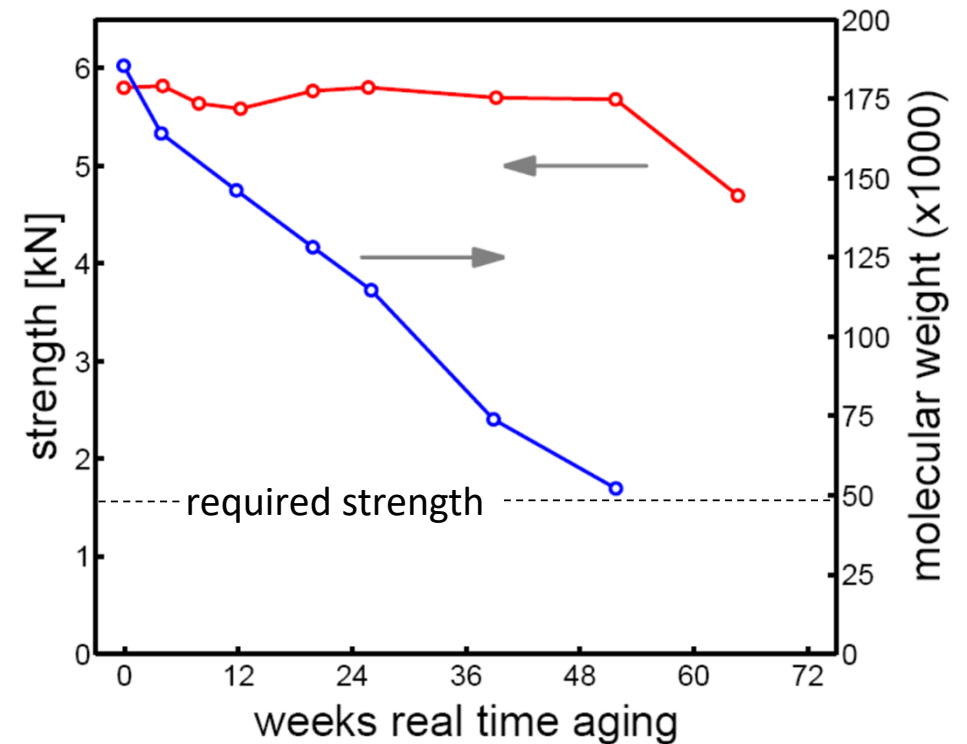
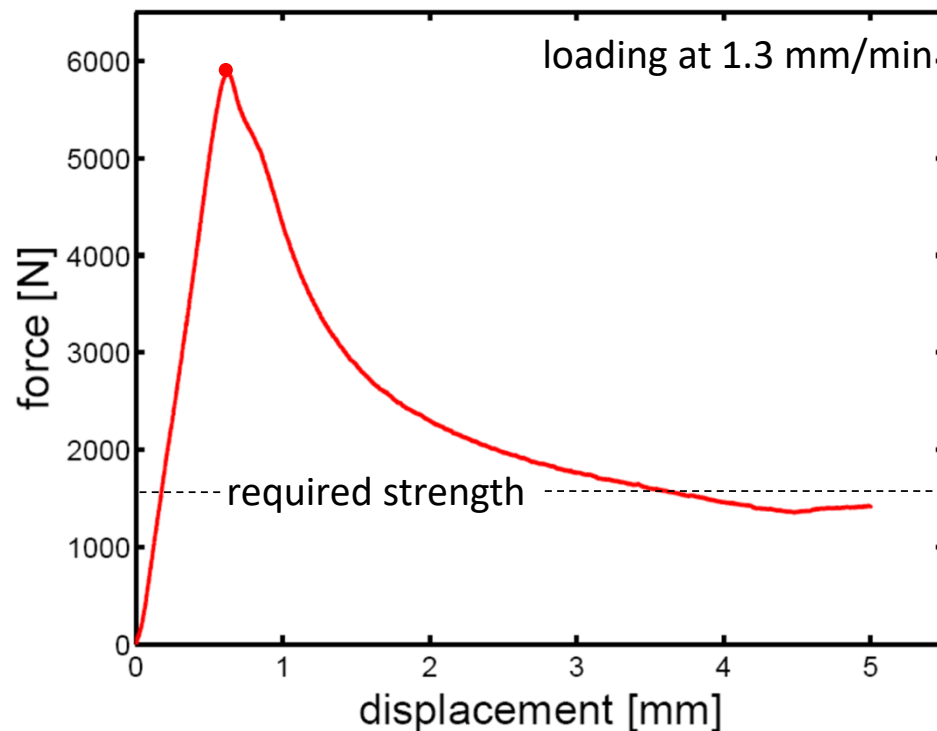
assessment of required strength



assessment of occurring loads using “instrumented” goat

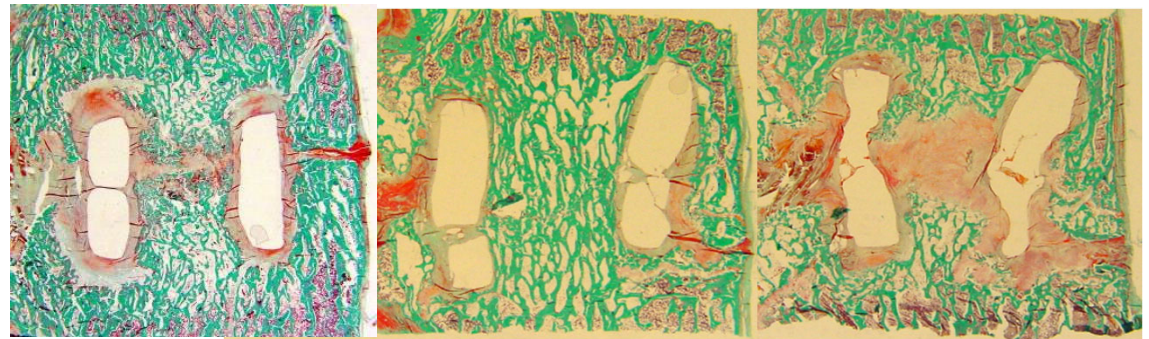


assessment of cage strength



for fusion, cage integrity required for at least 6 months: here achieved with **safety factor 6**

actual cage performance (in vivo)



3 months (non-fusion)

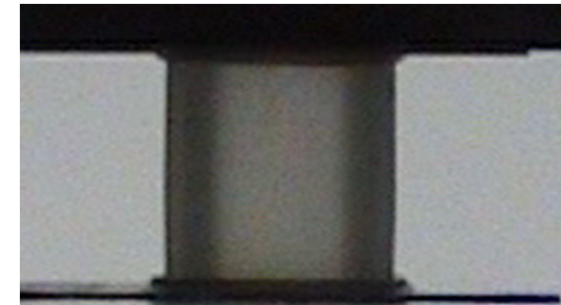
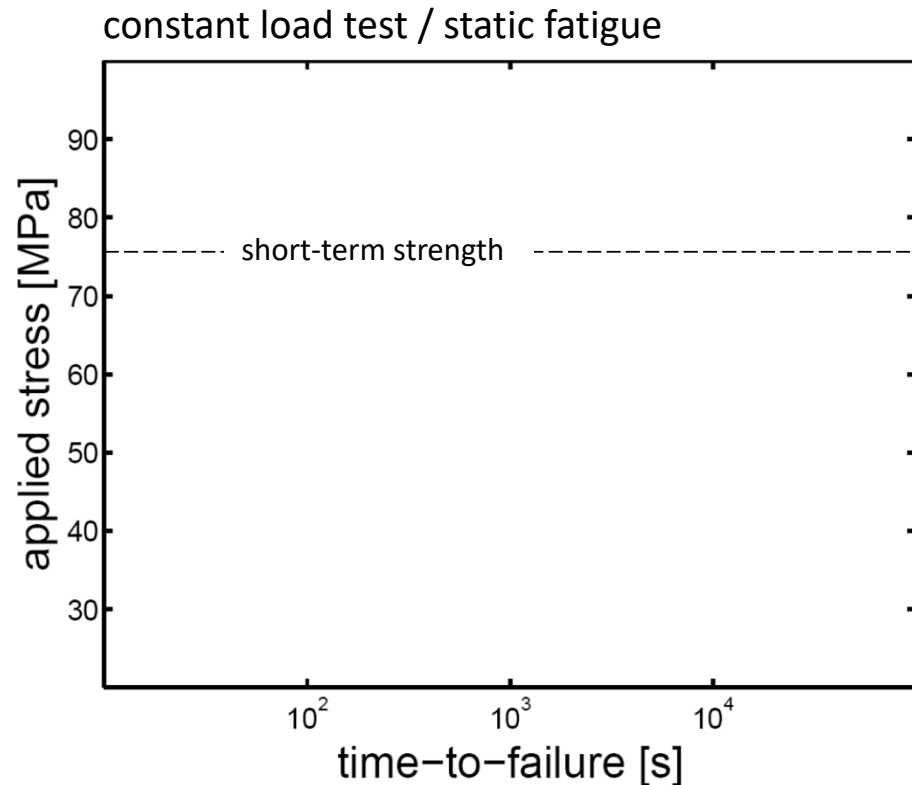
6 months (fusion)

6 months (non-fusion)

premature cage failure!!!

van Dijk M, Smit TH, Burger EH, Wuisman PI., Spine. 2002;27:2706–14.

failure under constant load: PLA



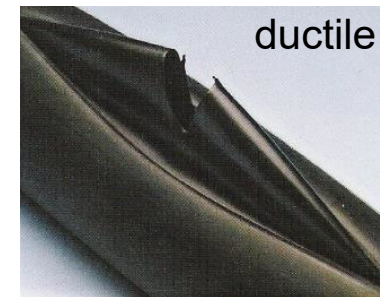
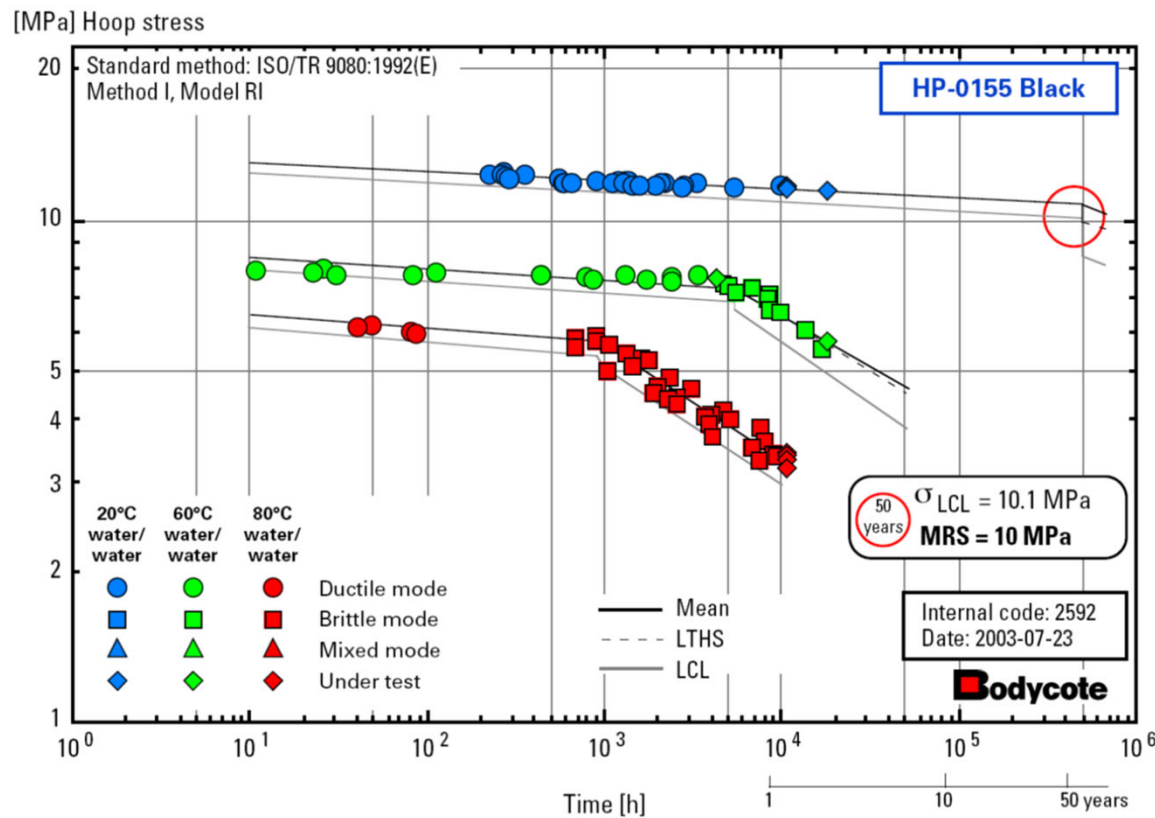
constant load 90% of
short term strength

if the actual load is lower than the short-term strength,
this does not imply that the material can sustain this load indefinitely

Fatigue and lifetime prediction strategies for polymers

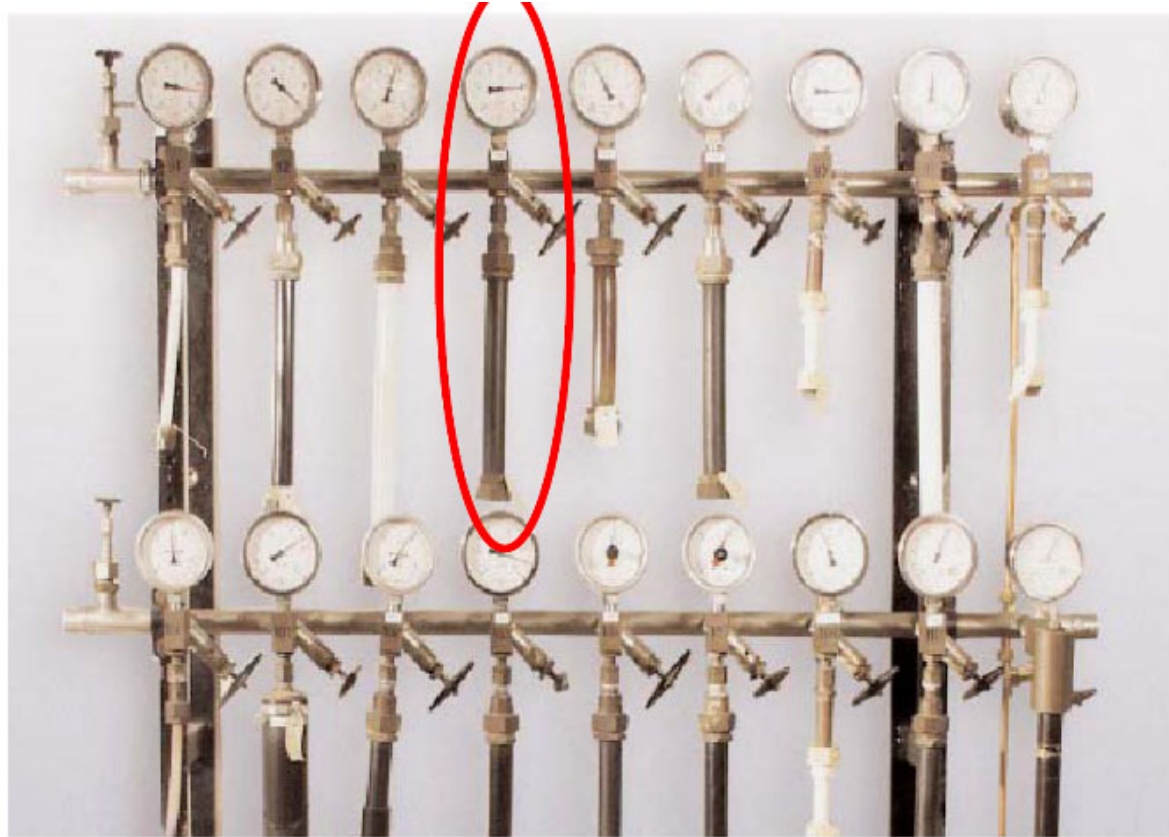
phenomenology & modelling

long-term hydrostatic testing



example: PE100 (HP-0155) (www.bodycotepolymer.com)
extrapolation using ISO 9080

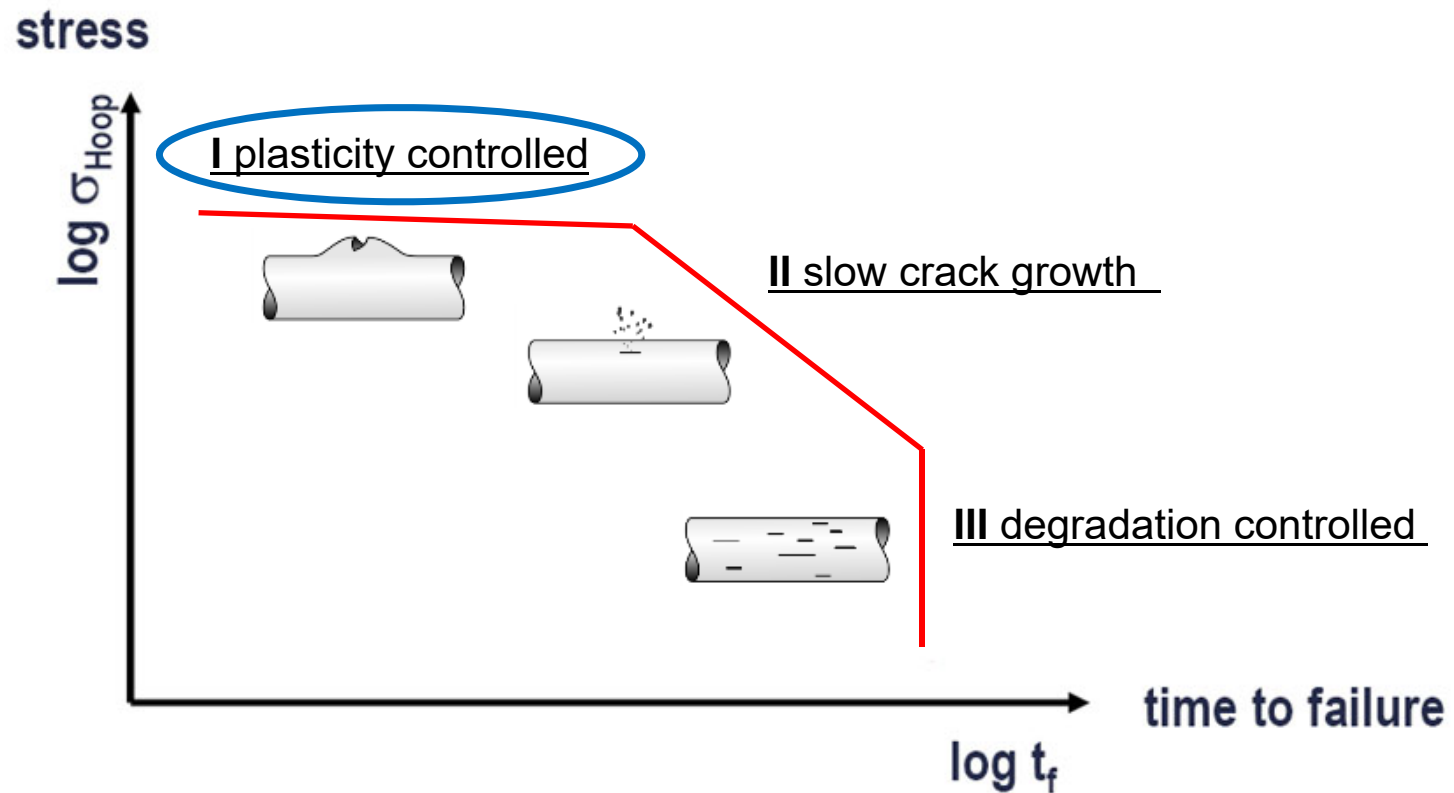
long-term testing: real-time loading?



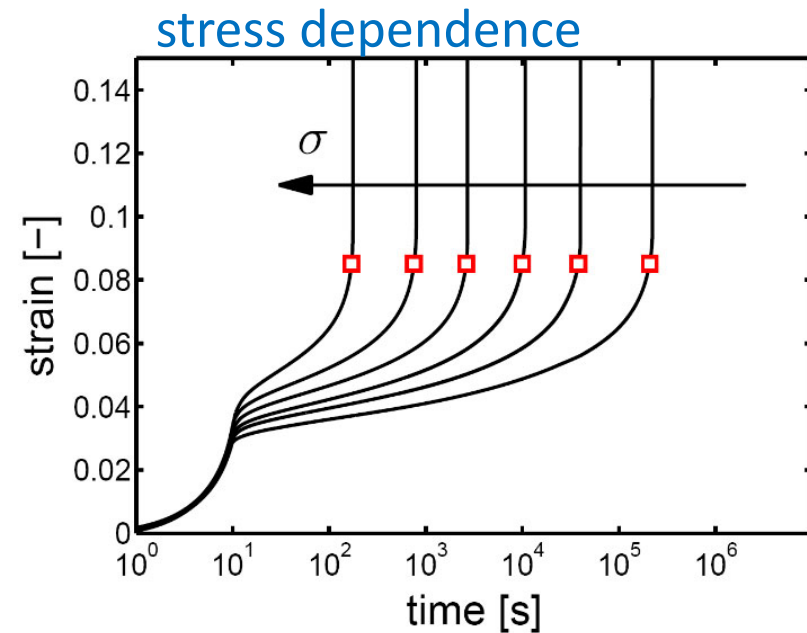
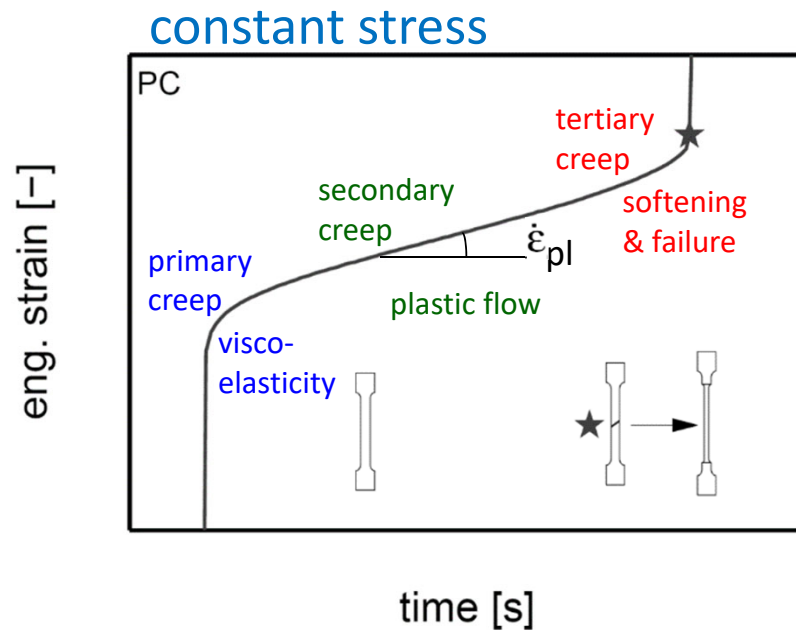
Hostalen GM 5010 under constant pressure since 1956

courtesy D. Lilge, Basell

long-term hydrostatic testing: failure modes

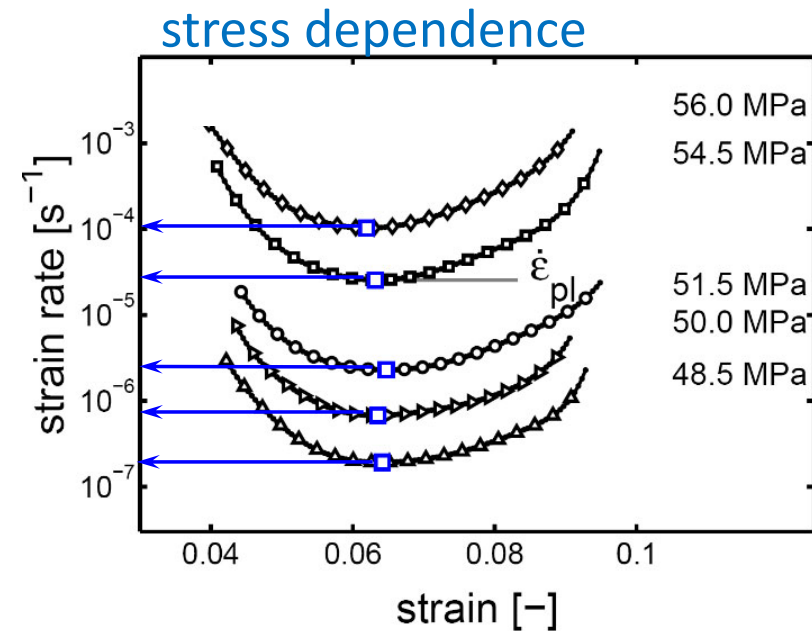
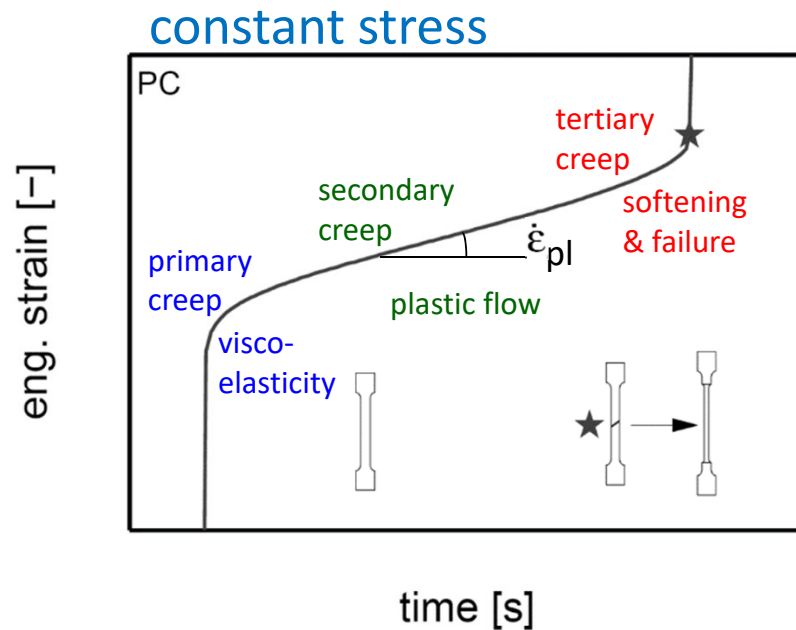


plasticity controlled failure: kinetics



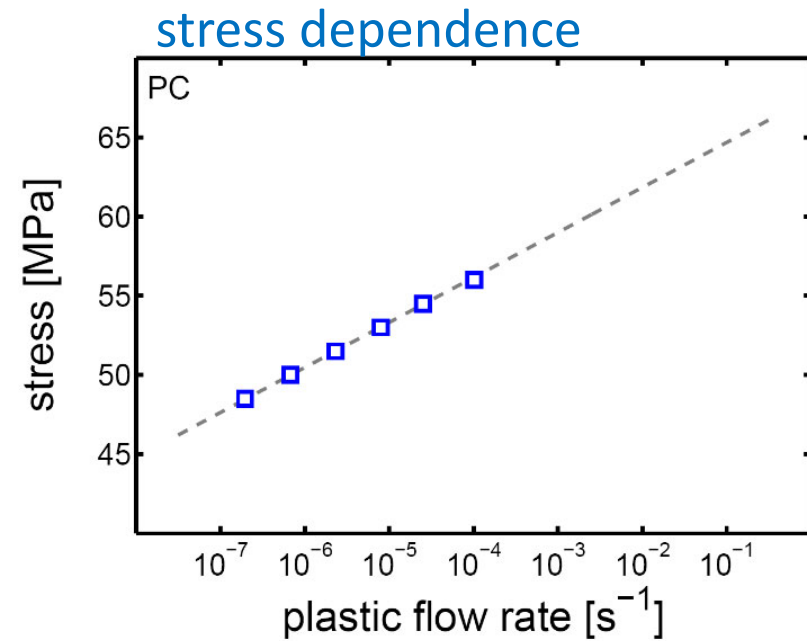
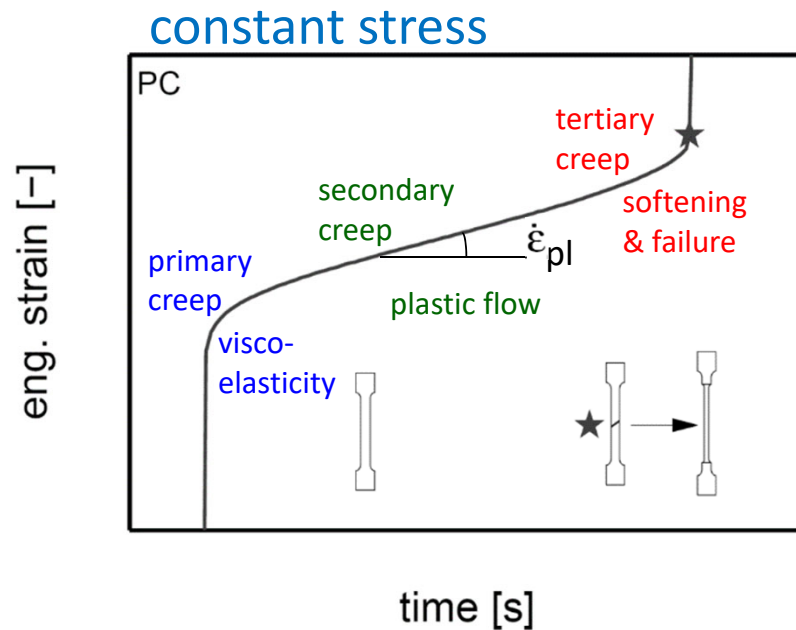
failure originates from accumulation of plastic strain

plasticity controlled failure: kinetics



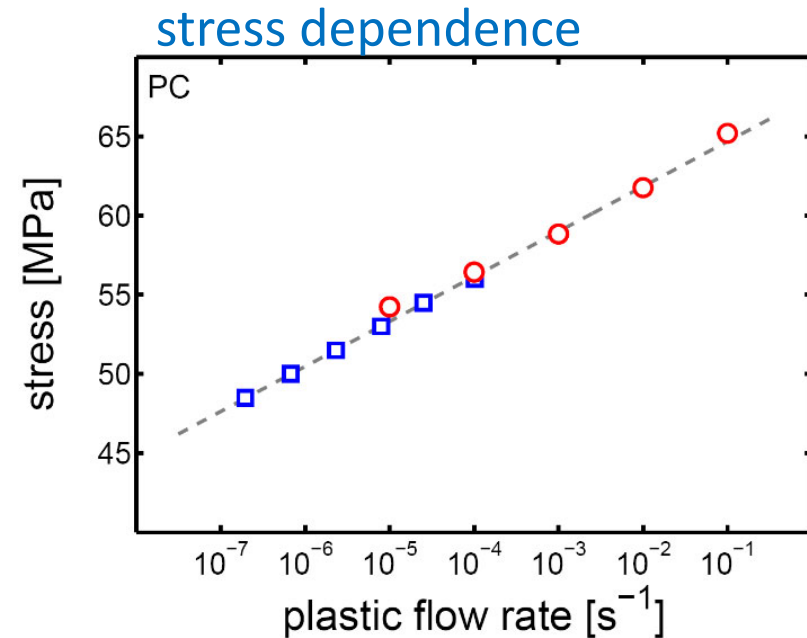
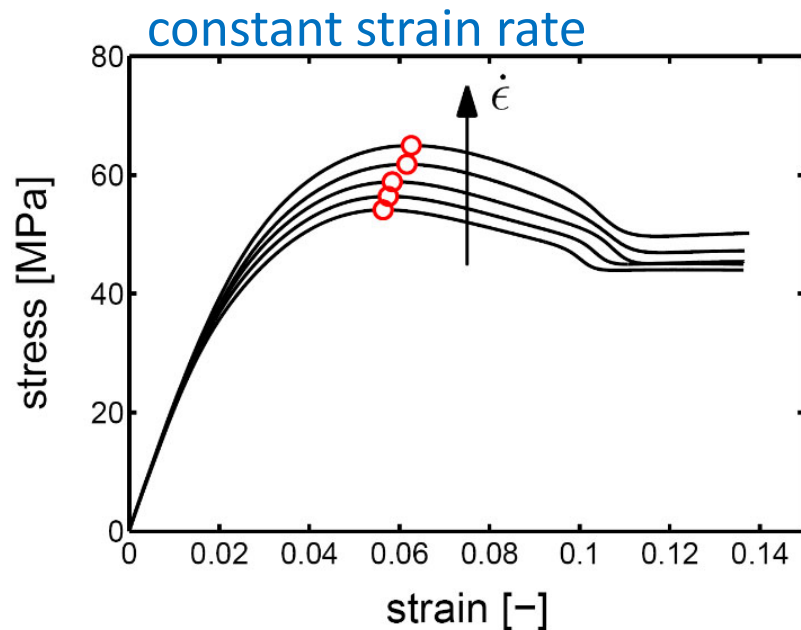
failure originates from accumulation of plastic strain

plasticity controlled failure: kinetics



failure originates from accumulation of plastic strain

plasticity controlled failure: kinetics



steady state reached in constant strain rate identical
to that reached in constant stress

Bauwens-Crowet et.al., 1974

plasticity controlled failure: kinetics

$$\dot{\epsilon}_{pl}(T, \sigma) = \underbrace{\dot{\epsilon}_0(S)}_{\text{changes in state}} \cdot \underbrace{\exp\left(-\frac{\Delta U}{RT}\right) \cdot \sinh\left(\frac{\sigma V^*}{kT}\right)}_{\text{deformation kinetics}}$$

changes in state

- prior thermal history
- progressive aging



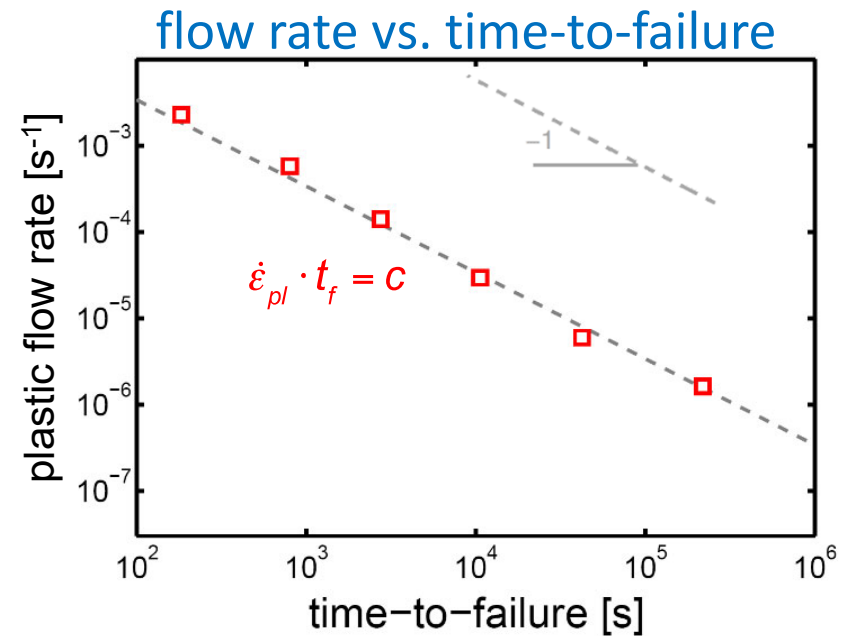
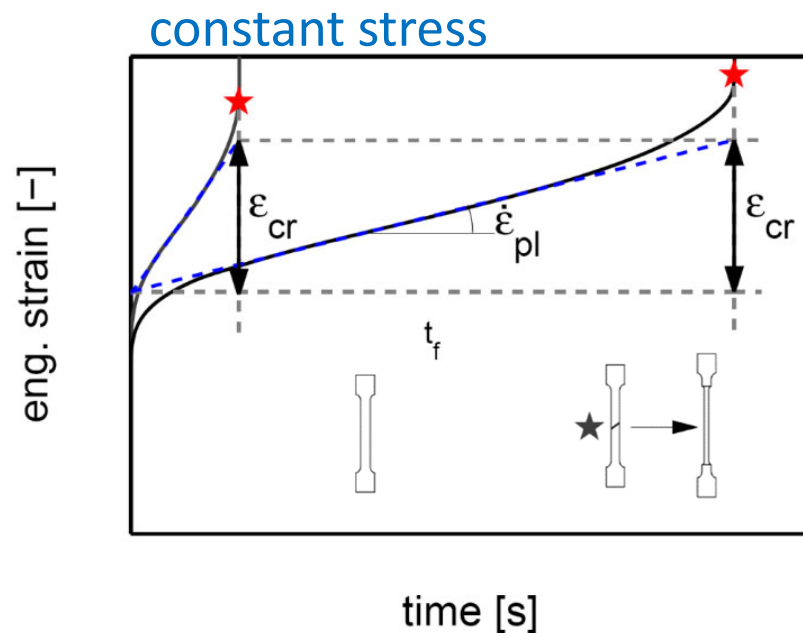
Henry Eyring

deformation kinetics

- stress-dependence
- temperature dependence

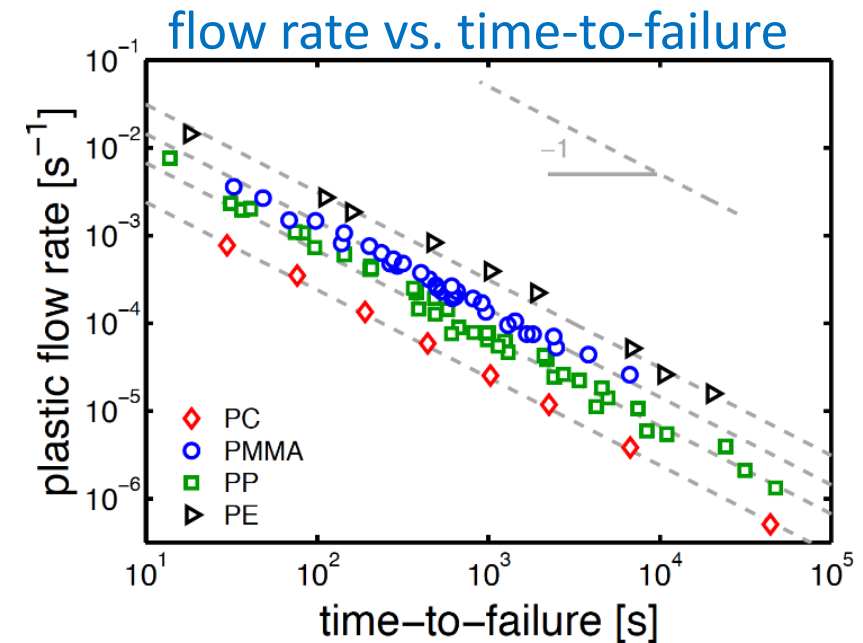
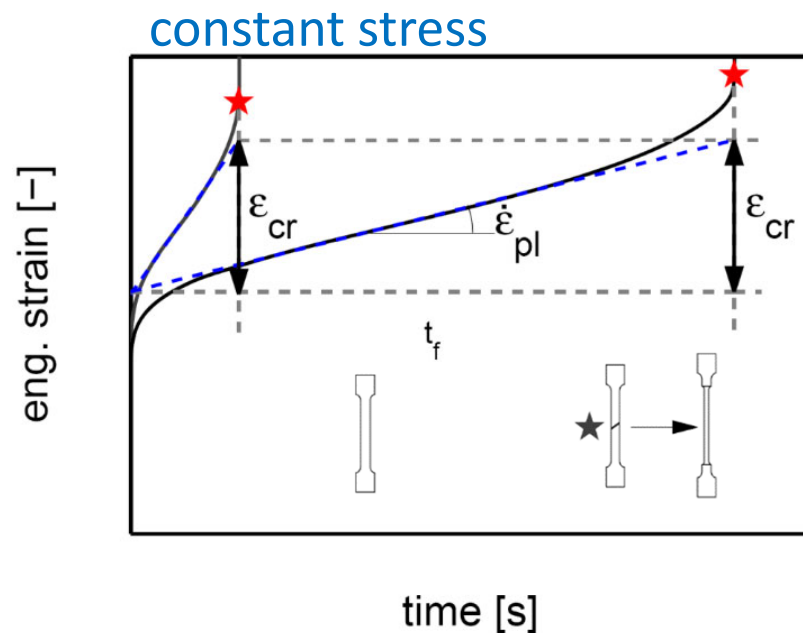
$\Delta U, V^*$ independent of thermodynamic state

life time prediction: concept of critical strain



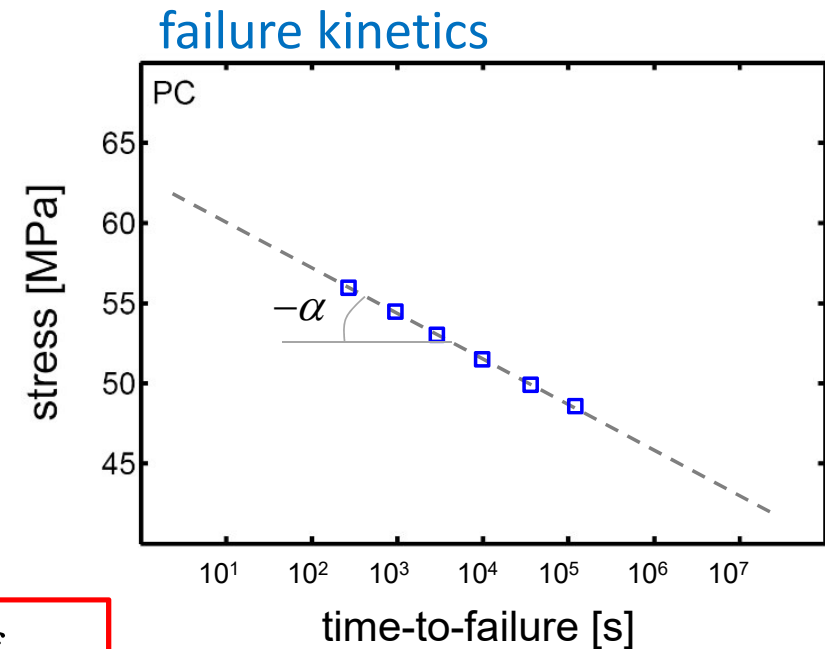
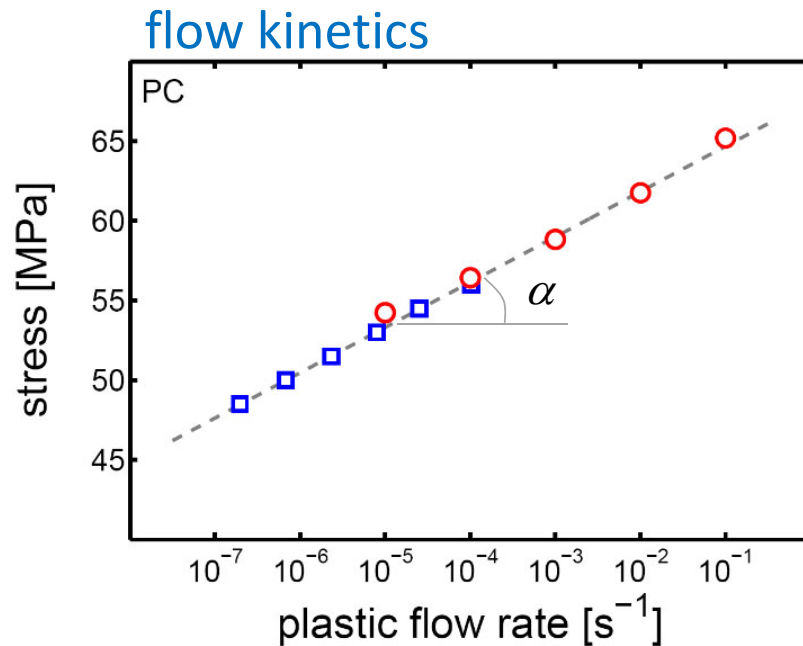
$$t_f(\sigma) = \frac{\epsilon_{cr}}{\dot{\epsilon}_{pl}(\sigma)}$$

life time prediction: concept of critical strain



$$t_f(\sigma) = \frac{\epsilon_{cr}}{\dot{\epsilon}_{pl}(\sigma)}$$

time-dependent failure: concept of critical strain



$$t_{fail} = \frac{\varepsilon_{cr}}{\dot{\varepsilon}_{pl}(T, \sigma)}$$

kinetics identical for both loading histories

time-dependent failure: concept of critical strain

accumulation of plastic strain proceeds at:

$$\dot{\varepsilon}_{pl}(T, \sigma) = \dot{\varepsilon}_0(S) \cdot \exp\left(-\frac{\Delta G}{RT}\right) \cdot \sinh\left(\frac{\sigma V^*}{kT}\right)$$

strain localization (failure) is triggered when a critical level of plastic strain has accumulated

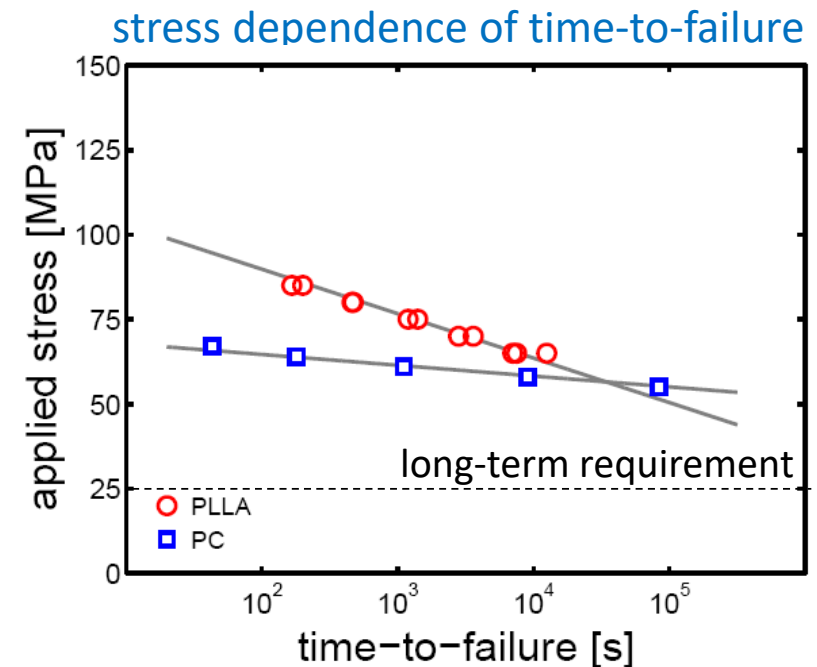
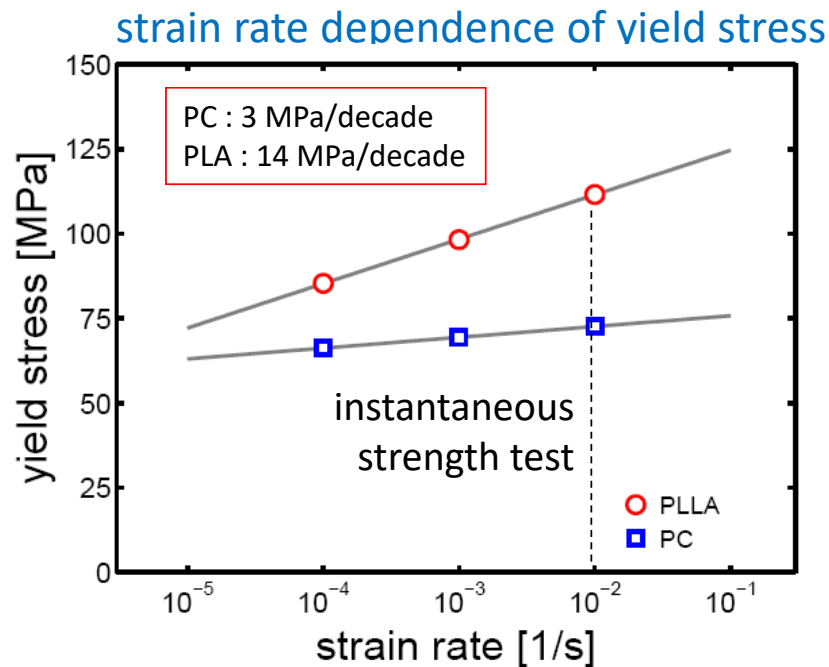
$$\varepsilon_{pl,acc} = \int_0^t \dot{\varepsilon}_{pl}(T, \sigma) dt'$$

failure if: $\varepsilon_{pl,acc} \geq \varepsilon_{cr}$

constant loading:

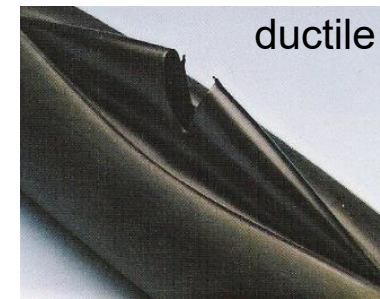
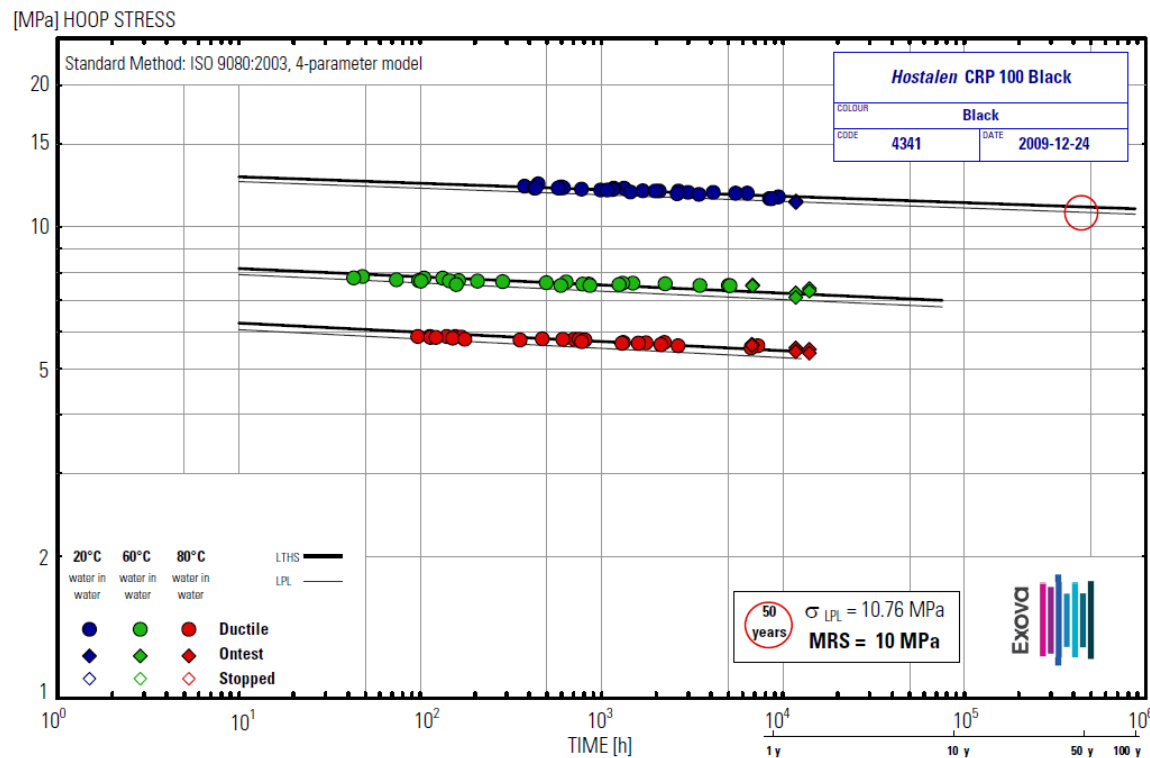
$$t_{fail} = \frac{\varepsilon_{cr}}{\dot{\varepsilon}_{pl}(T, \sigma)}$$

time-dependent failure: importance of kinetics



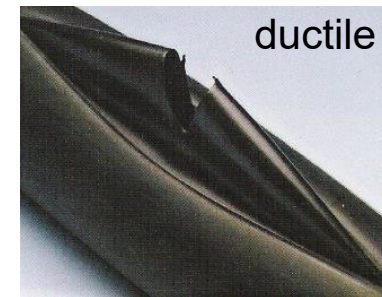
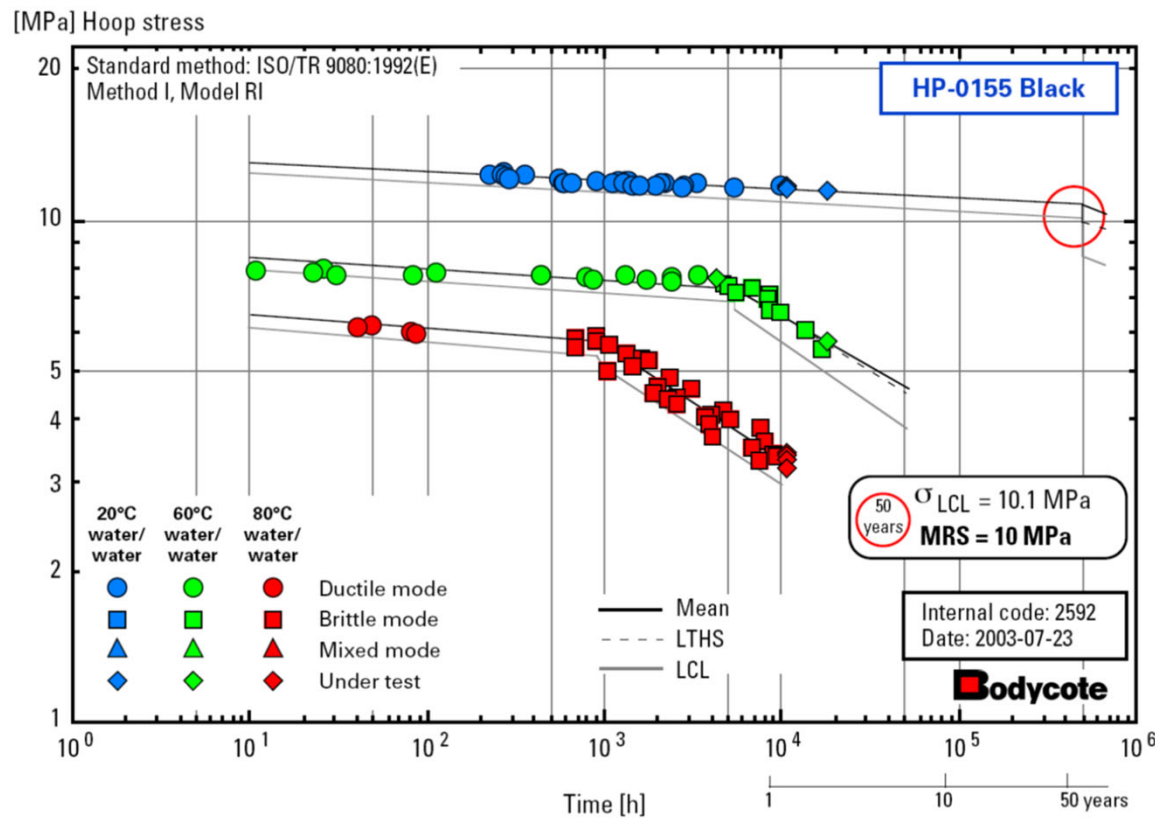
- PLA appears significantly stronger than e.g. PC when tested under constant strain rates
- under constant stress, however, PC outperforms PLA already within a day!

long-term hydrostatic testing



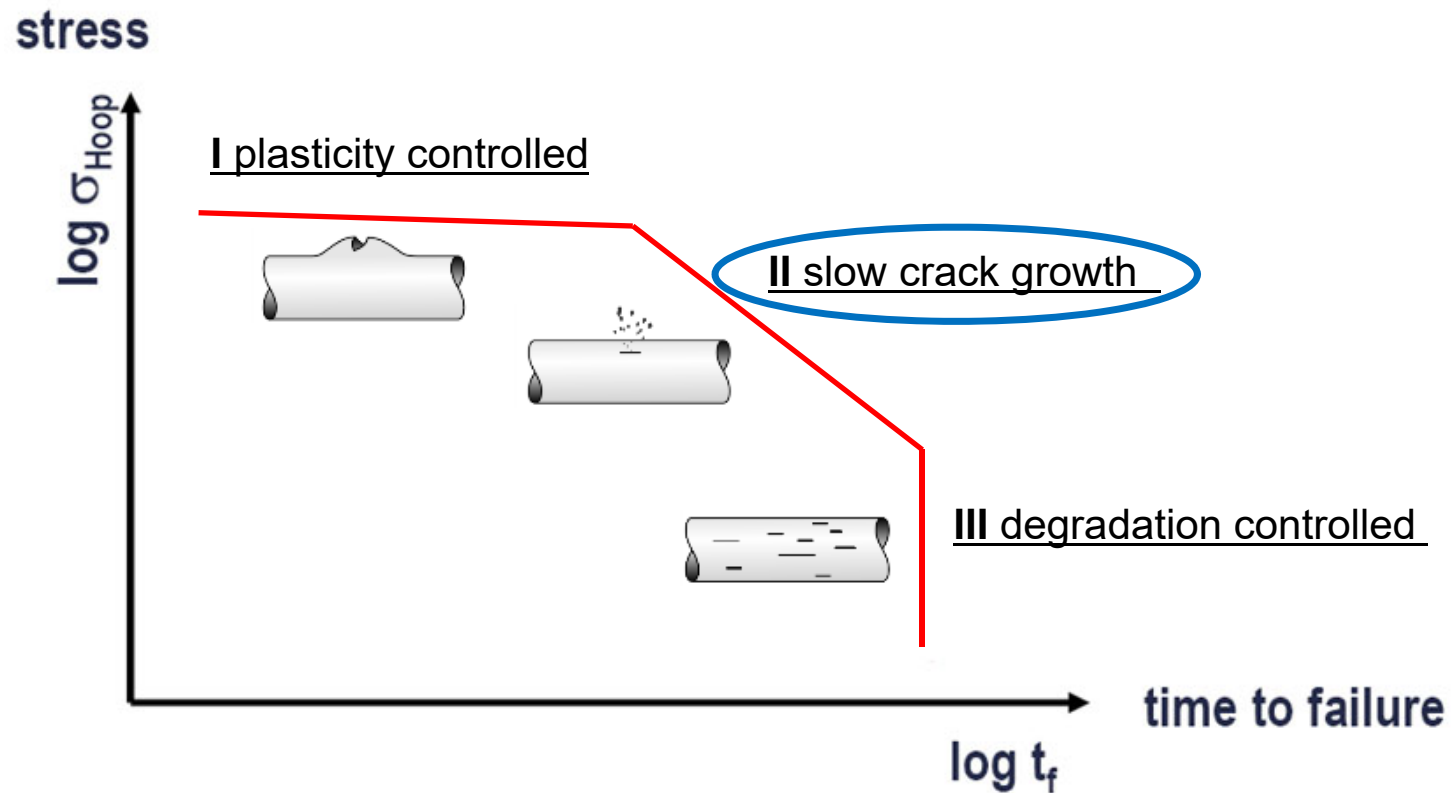
example: PE100 (HP-0155) (www.bodycotepolymer.com)
extrapolation using ISO 9080

long-term hydrostatic testing



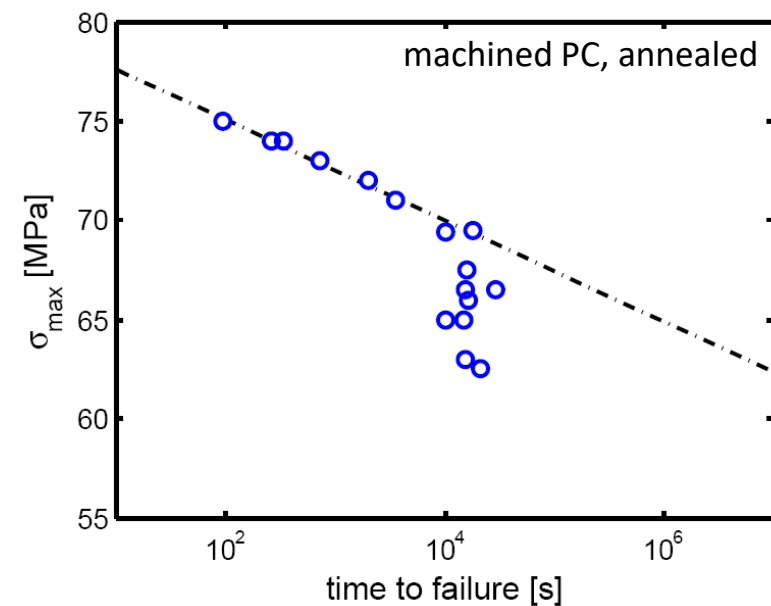
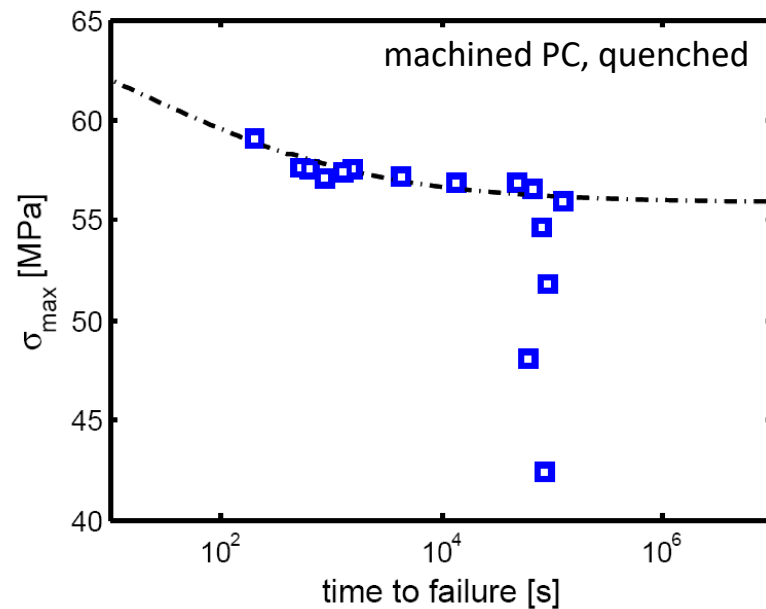
example: PE100 (CRP-100) (www.bodycotepolymer.com)
extrapolation using ISO 1167

long-term hydrostatic testing: failure modes



brittle failure: crack growth!

fatigue loading, 0.1 Hz, $\sigma_{\min} = 2$ MPa



change in failure mode and kinetics!

growth and failure of a crack that is initially present

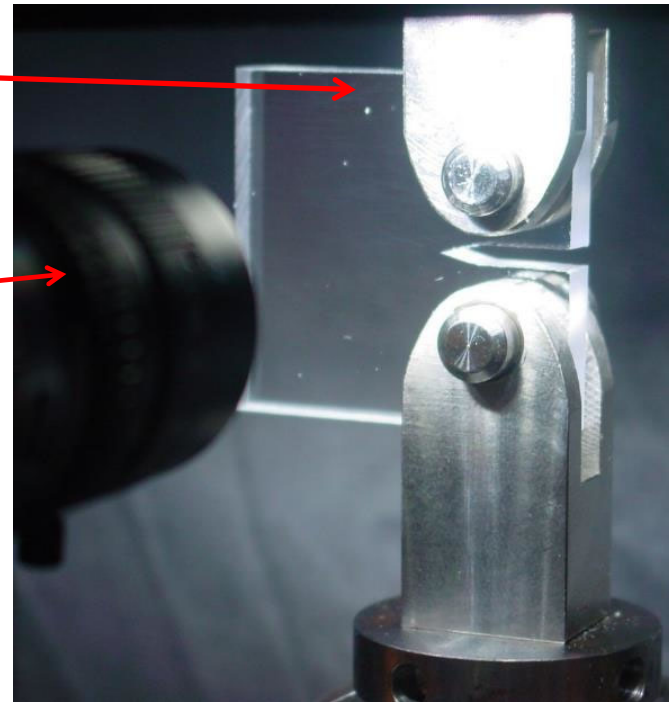
brittle failure: crack growth measurement

servo-hydraulic tensile tester in force control



pre-cracked
CT specimen

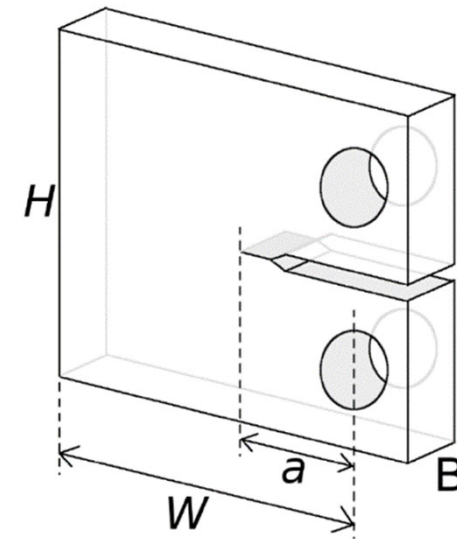
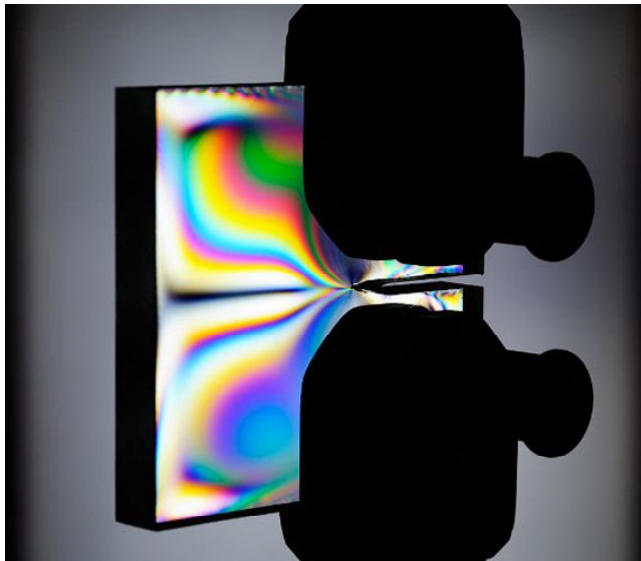
camera



crack length determined by image processing.

brittle failure: crack growth measurement

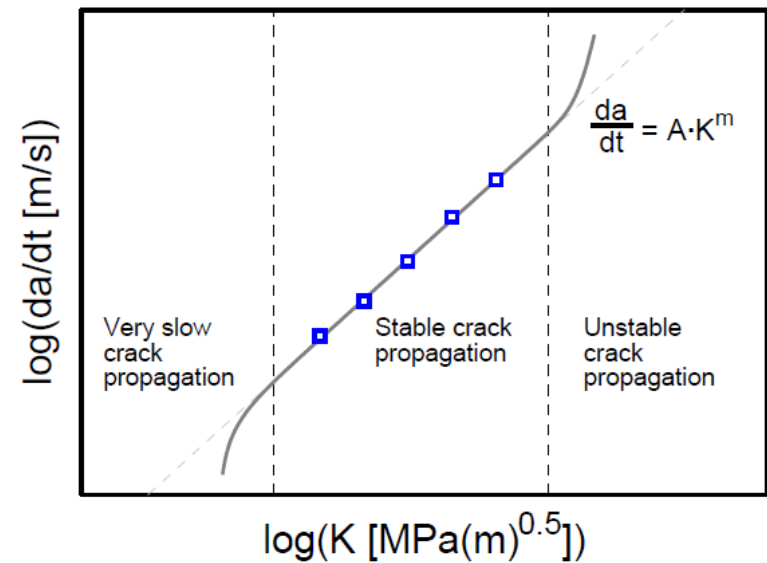
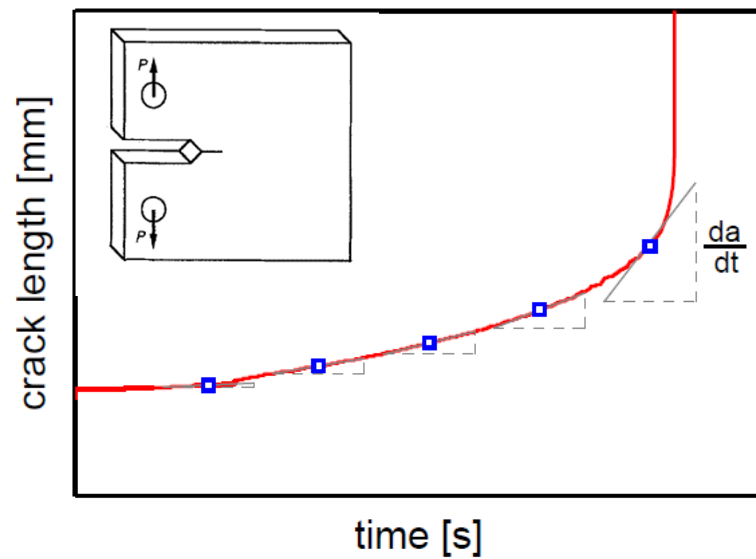
Compact Tension (CT-)specimen



$$K_I = \frac{F}{B\sqrt{W}} \left[0.886 + 4.64 \left(\frac{a}{W} \right) - 13.32 \left(\frac{a}{W} \right)^2 + 14.72 \left(\frac{a}{W} \right)^3 - 5.6 \left(\frac{a}{W} \right)^4 \right]$$

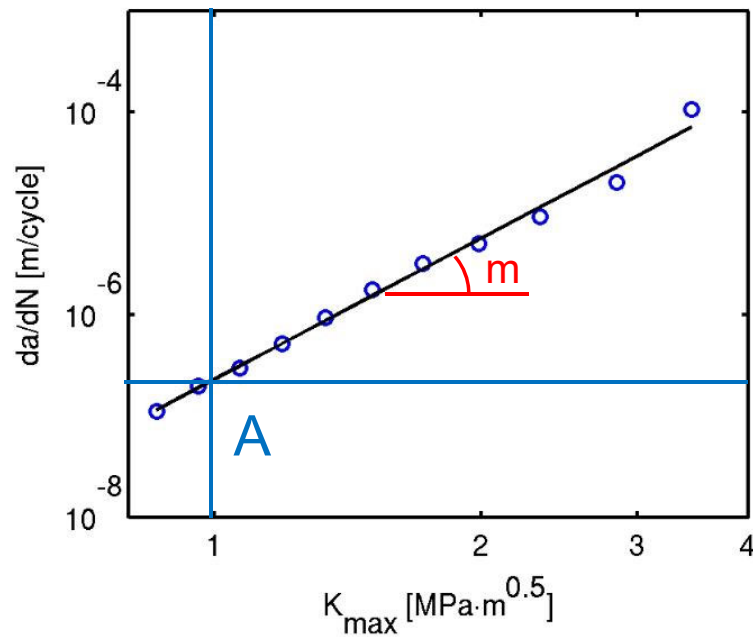
mode I slow crack growth

crack growth monitored under static load



"Paris Law": $\frac{da}{dt} = AK^m$

crack propagation rate



Paris law:

$$\frac{da}{dN} = A K_{max}^m$$

The parameter A is labeled "value at $K_{max} = 1$ ". The parameter m is labeled "slope".

slow crack growth: lifetime prediction

Paris law:

$$\frac{da}{dt} = AK^m = A(Y\sigma\sqrt{\pi a})^m$$

integrate over crack length:

$$t_f = \int_0^{t_f} dt = \int_{a_i}^{a_f} \frac{1}{A(Y\sigma\sqrt{\pi a})^m} da$$

lifetime versus stress:

$$t_f = \left(\frac{1}{A} \int_{a_i}^{a_f} \frac{1}{(Y\sqrt{\pi a})^m} da \right) \cdot \sigma^{-m}$$

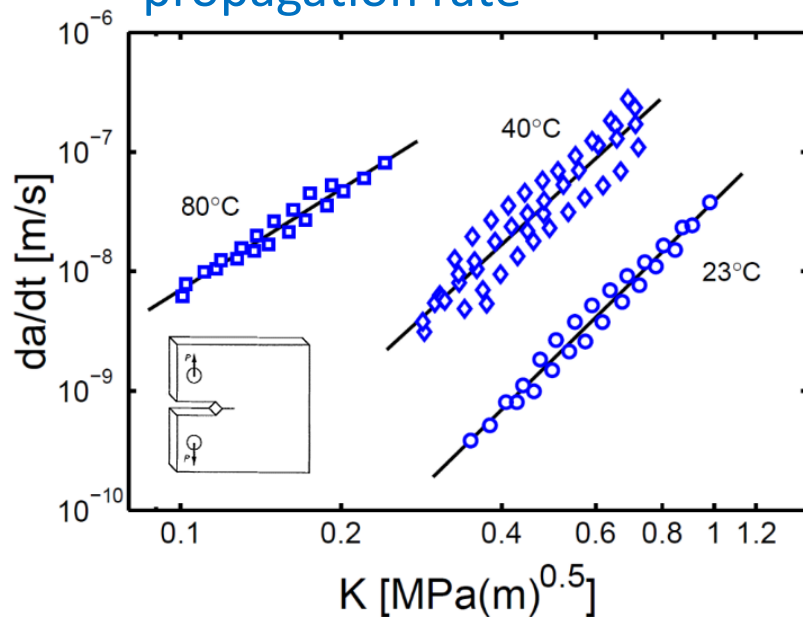
stress versus lifetime:

$$\sigma = \left(\frac{1}{A} \int_{a_i}^{a_f} \frac{1}{(Y\sqrt{\pi a})^m} da \right)^{\frac{1}{m}} \cdot t_f^{\frac{1}{m}}$$

$$\sigma = c_f \cdot t_f^{\frac{1}{m}}$$

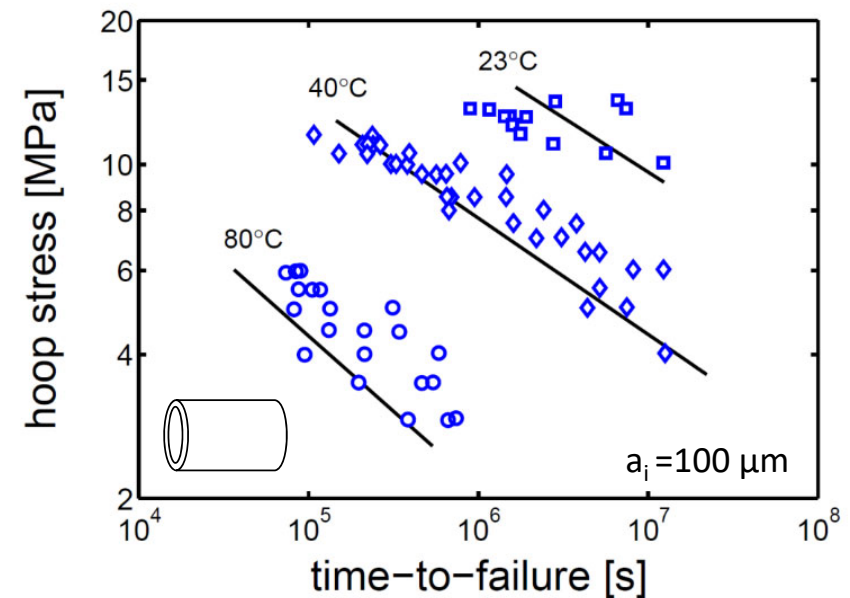
slow crack growth: lifetime prediction

propagation rate



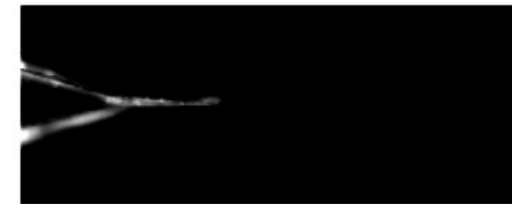
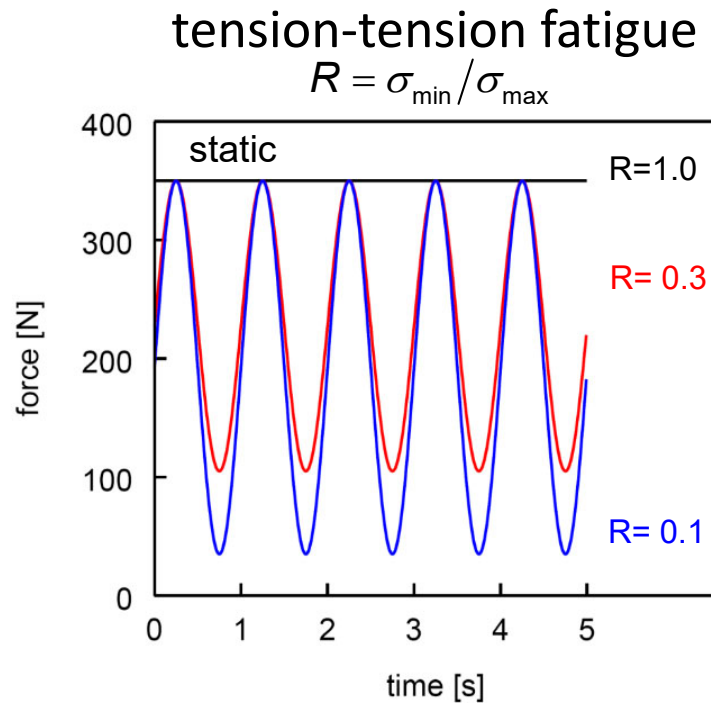
$$\frac{da}{dt} = AK^m$$

lifetime



$$\sigma = c_f \cdot t_f^{-1/m}$$

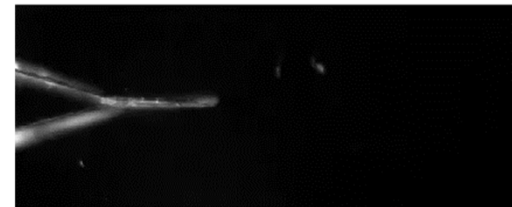
slow crack growth: static versus cyclic loading



R=1.0



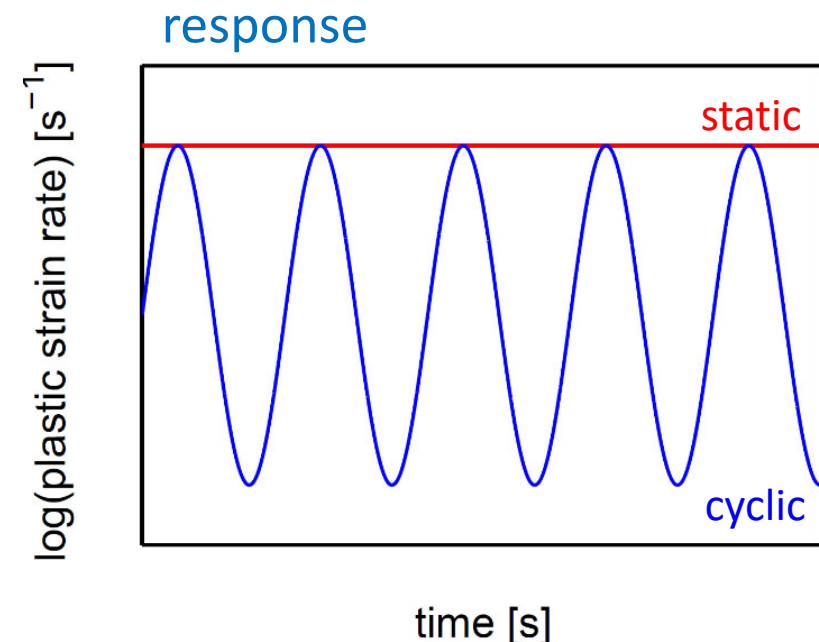
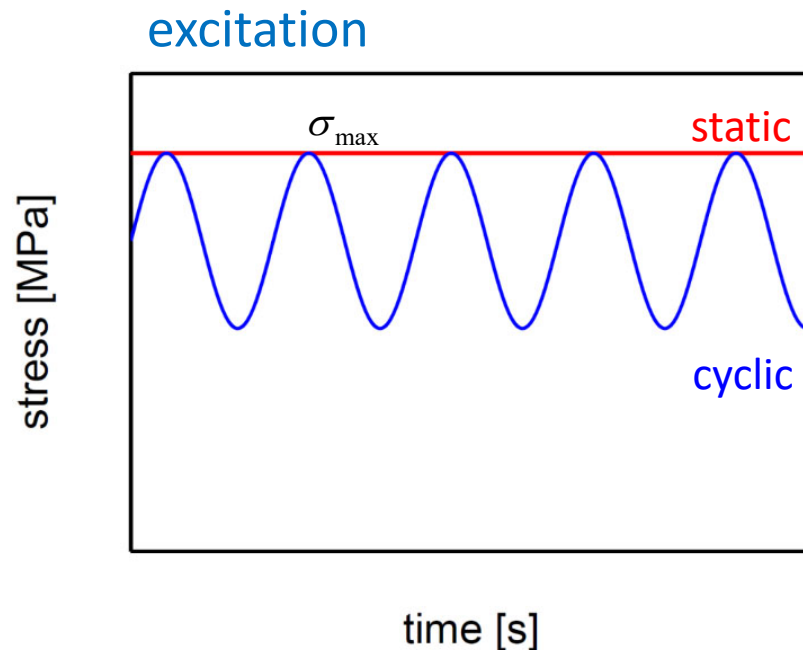
R= 0.3



R= 0.1

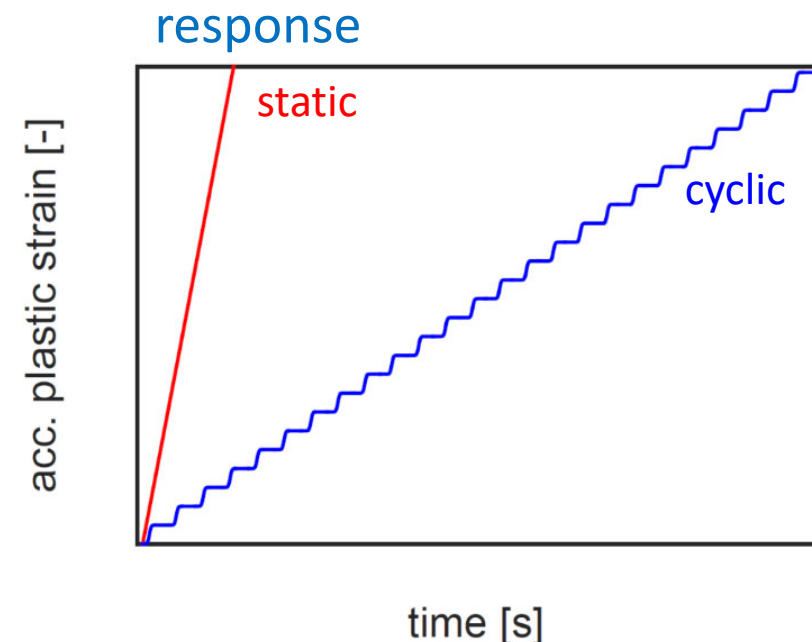
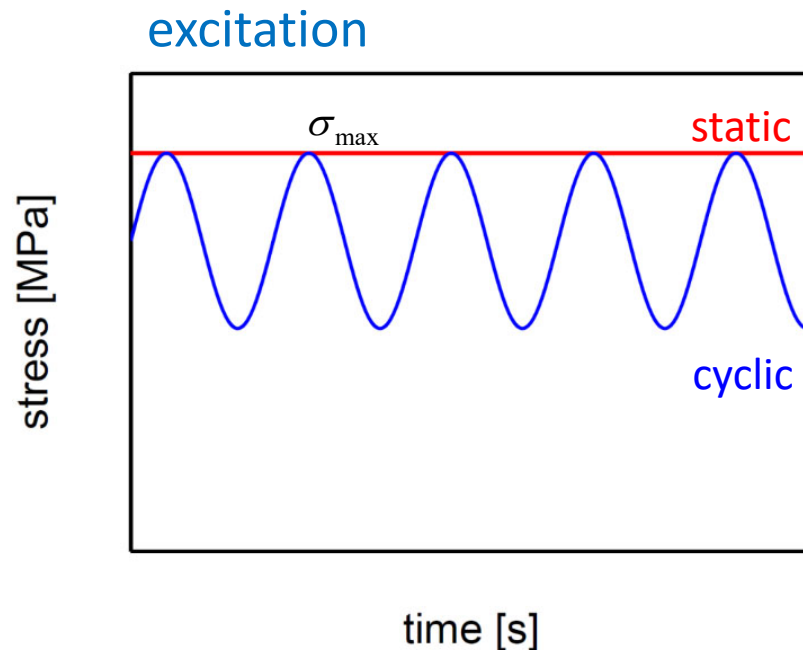
strong acceleration of crack growth in cyclic loading!

plasticity-controlled failure: cyclic loading



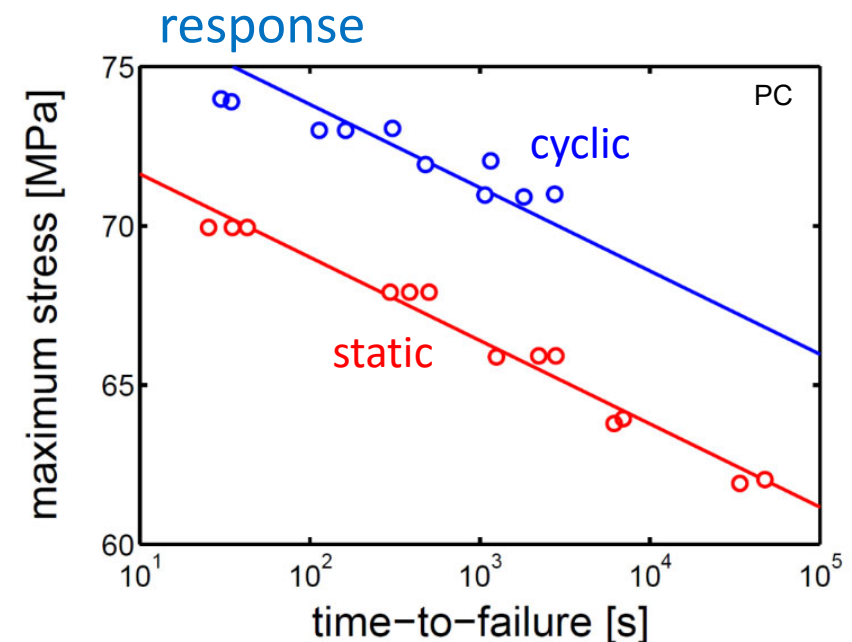
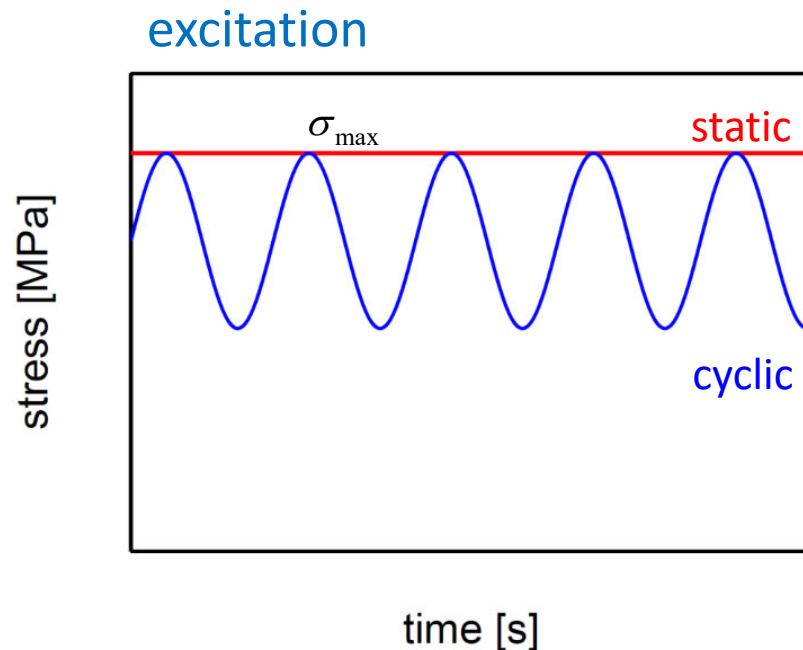
$$\varepsilon_{pl,acc} = \int_0^t \dot{\varepsilon}_{pl}(T, \sigma) dt' \quad \text{failure occurs when:} \quad \varepsilon_{pl,acc} \geq \varepsilon_{cr}$$

plasticity-controlled failure: cyclic loading



$$\varepsilon_{pl,acc} = \int_0^t \dot{\varepsilon}_{pl}(T, \sigma) dt' \quad \text{failure occurs when:} \quad \varepsilon_{pl,acc} \geq \varepsilon_{cr}$$

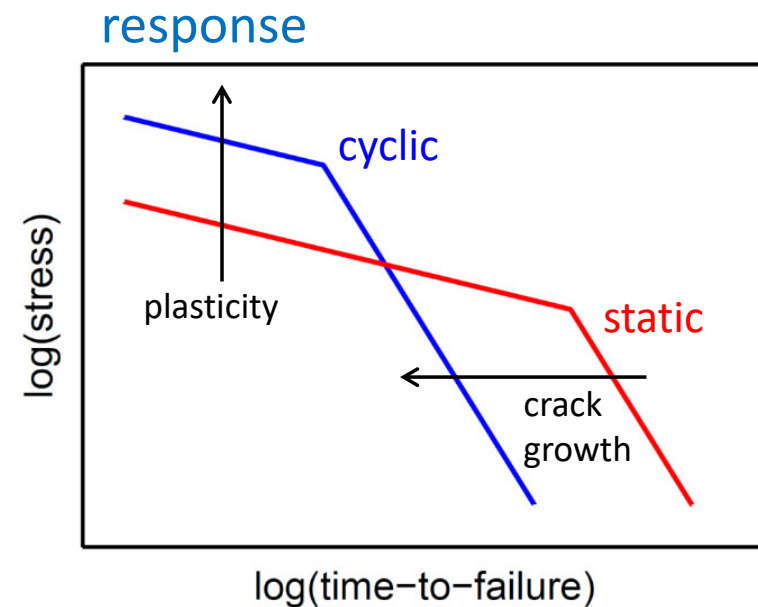
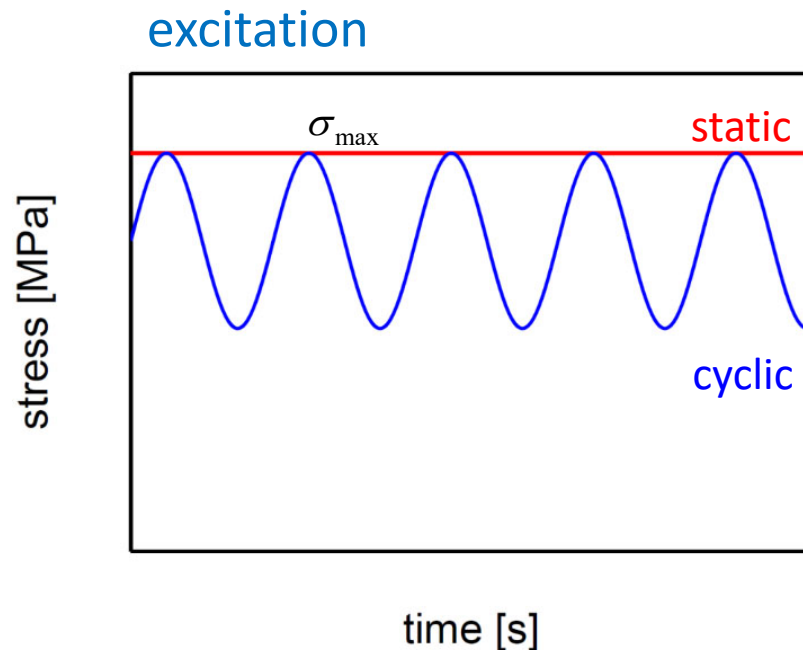
plasticity-controlled failure: cyclic loading



lifetime improves as a result of decreasing rate of plastic strain accumulation

cyclic load has longest time-to-failure

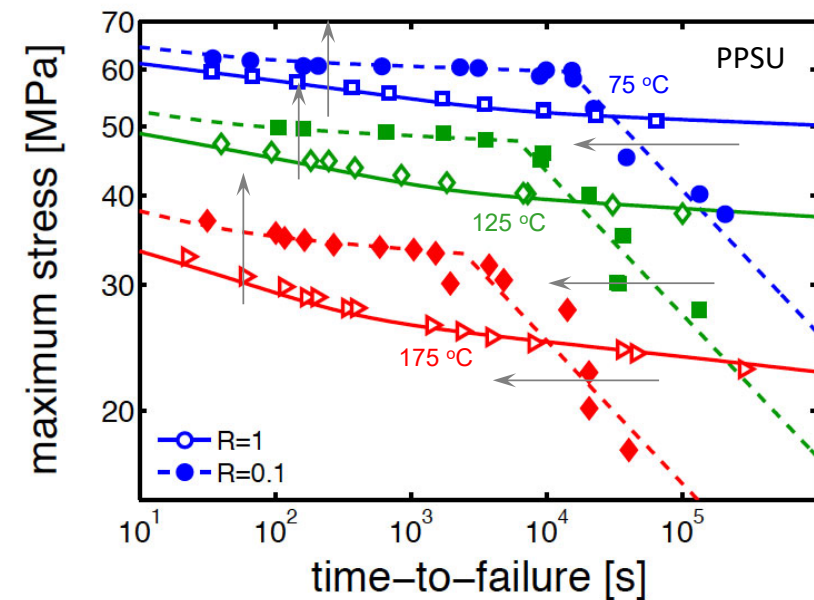
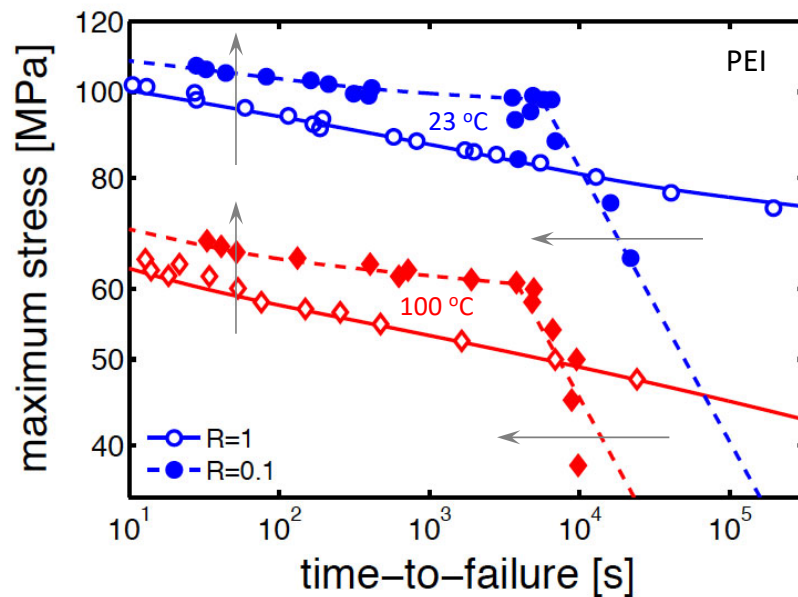
cyclic vs. static loading: time-to-failure



cyclic loading has a different effect on both failure modes!

cyclic vs. static loading: smooth bars

glassy polymers



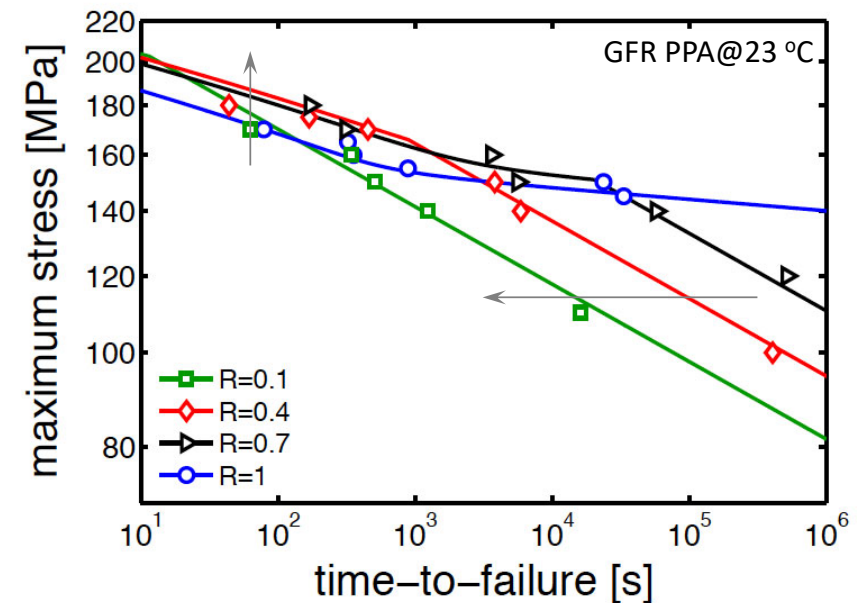
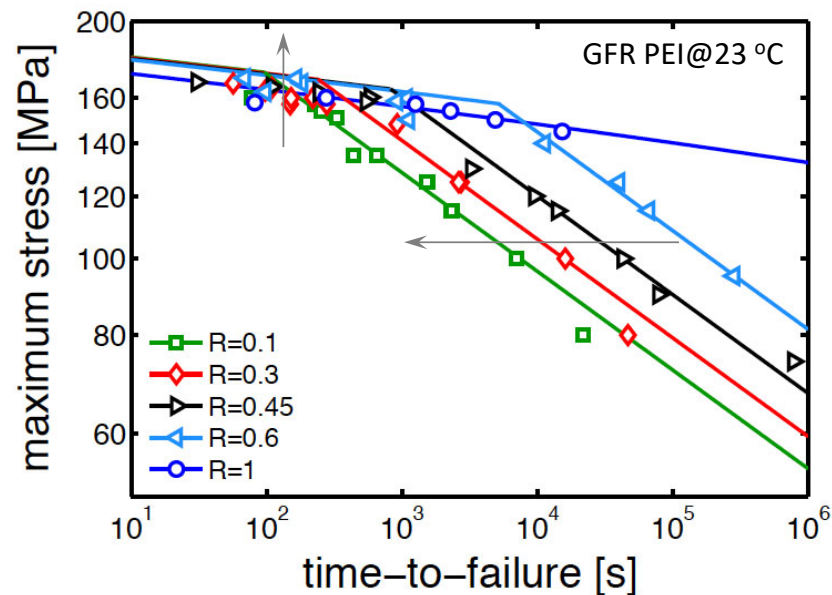
plasticity-controlled failure



slow crack growth

cyclic vs. static loading: smooth bars

fibre-reinforced systems

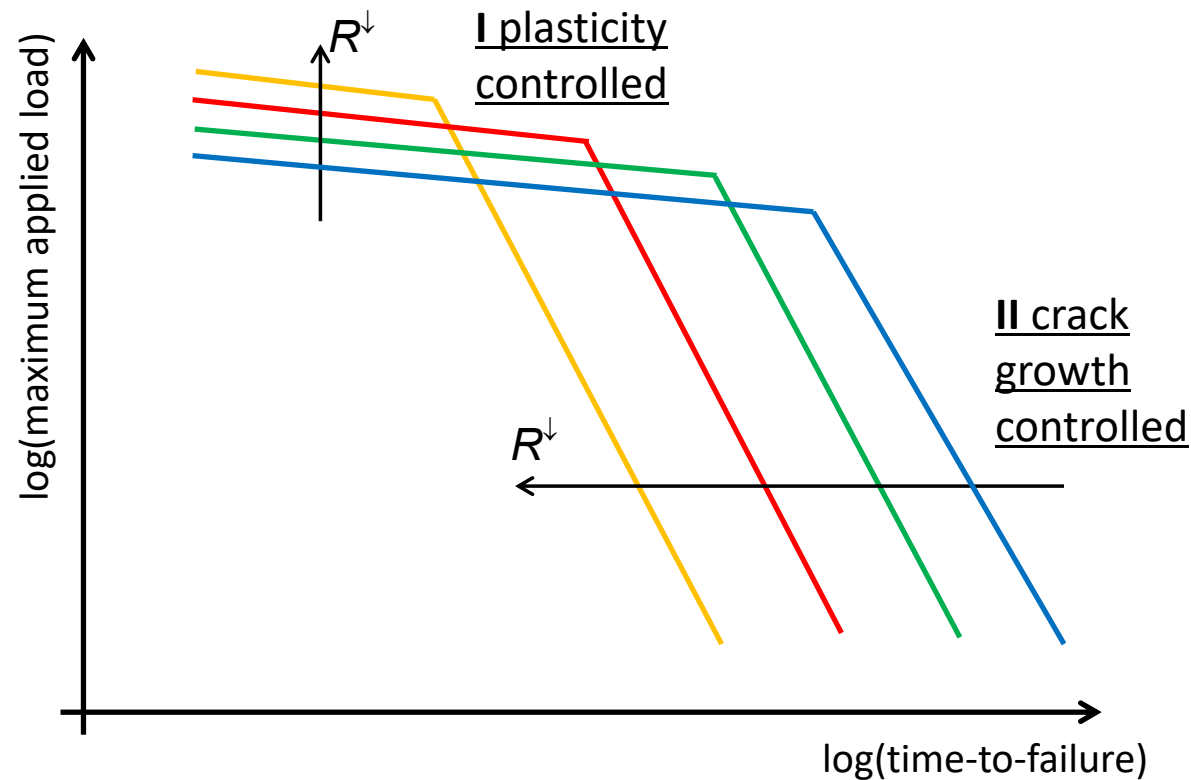


plasticity-controlled failure



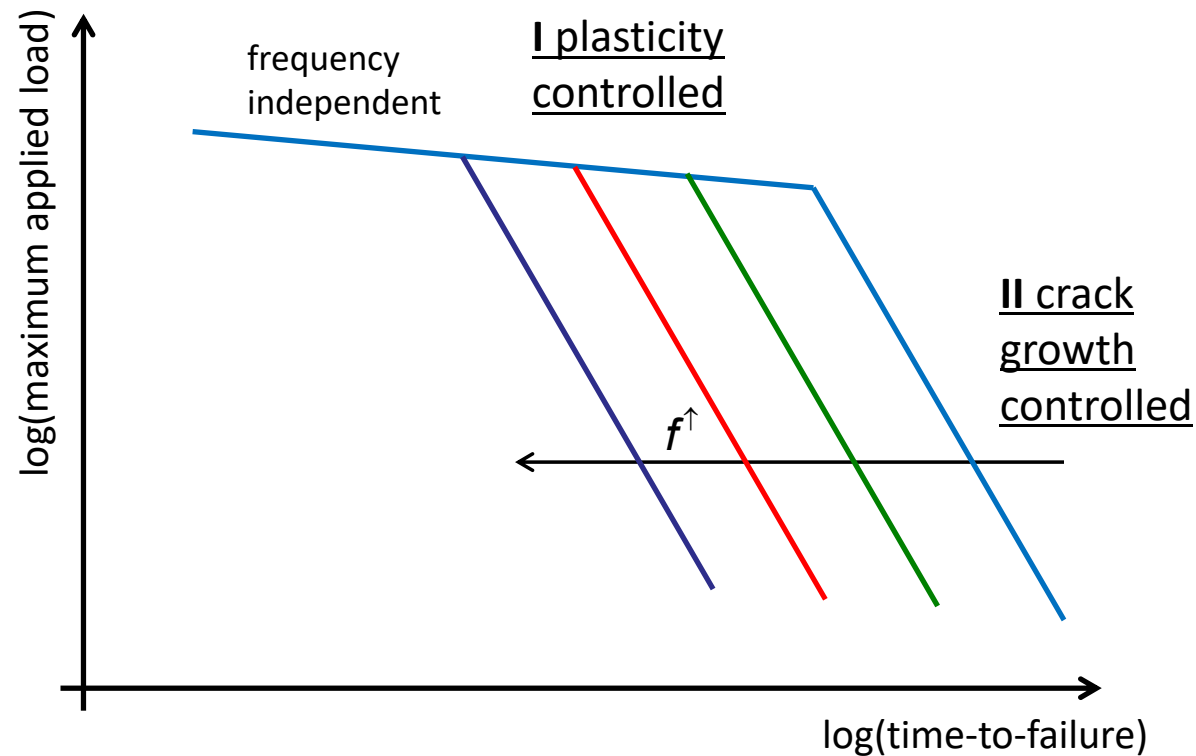
slow crack growth

trends time-to-failure: amplitude ratio R



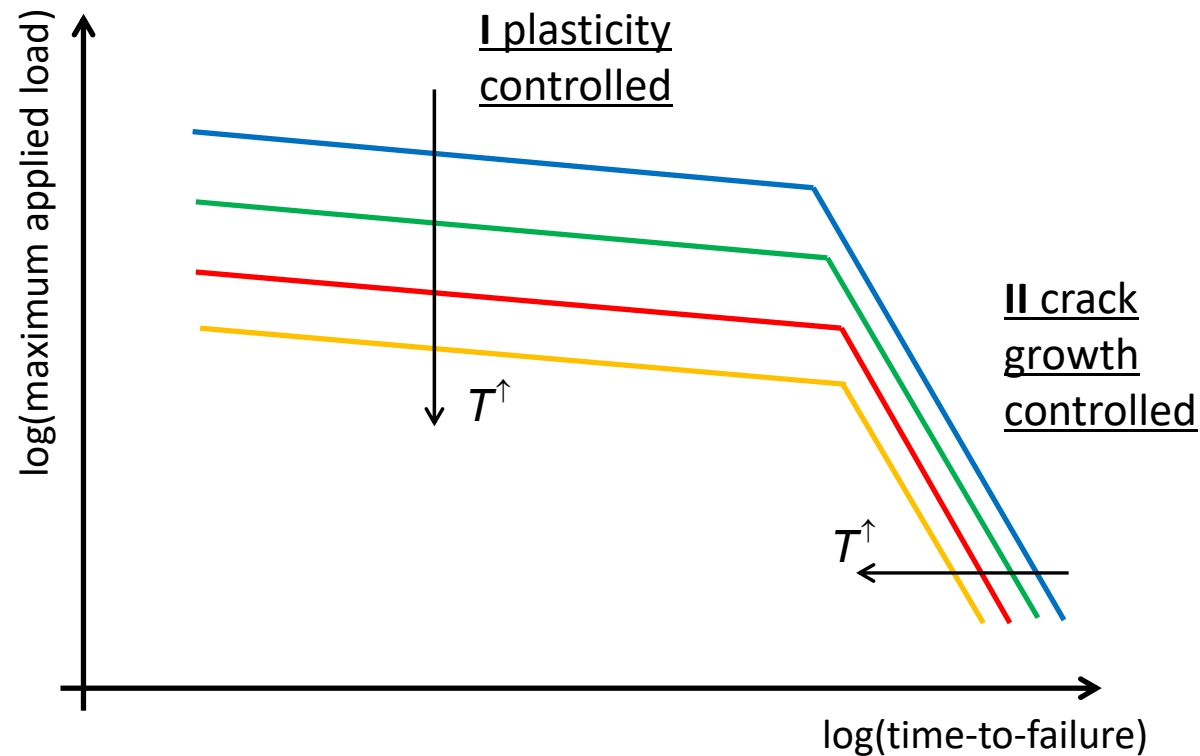
plasticity controlled failure: lifetime increases with decreasing R
slow crack growth: lifetime decreases with decreasing R

trends time-to-failure: frequency



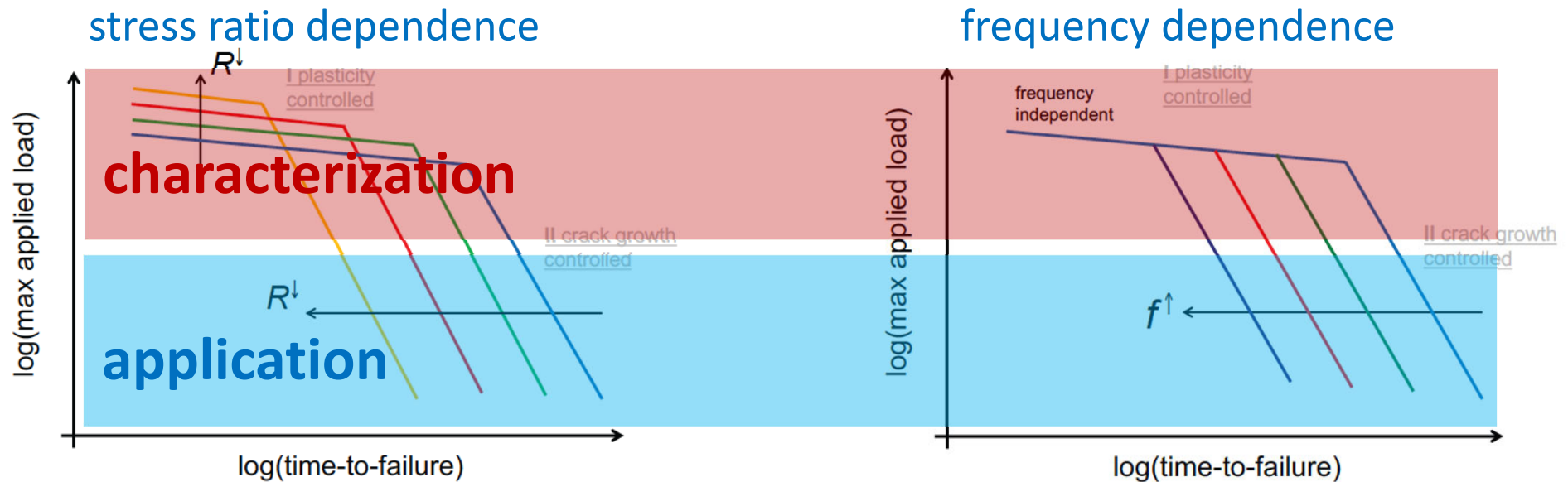
plasticity controlled failure: lifetime independent of frequency
slow crack growth: lifetime decreases with increasing frequency

trends time-to-failure: temperature



plasticity controlled failure: strong temperature dependence
slow crack growth : weak temperature dependence

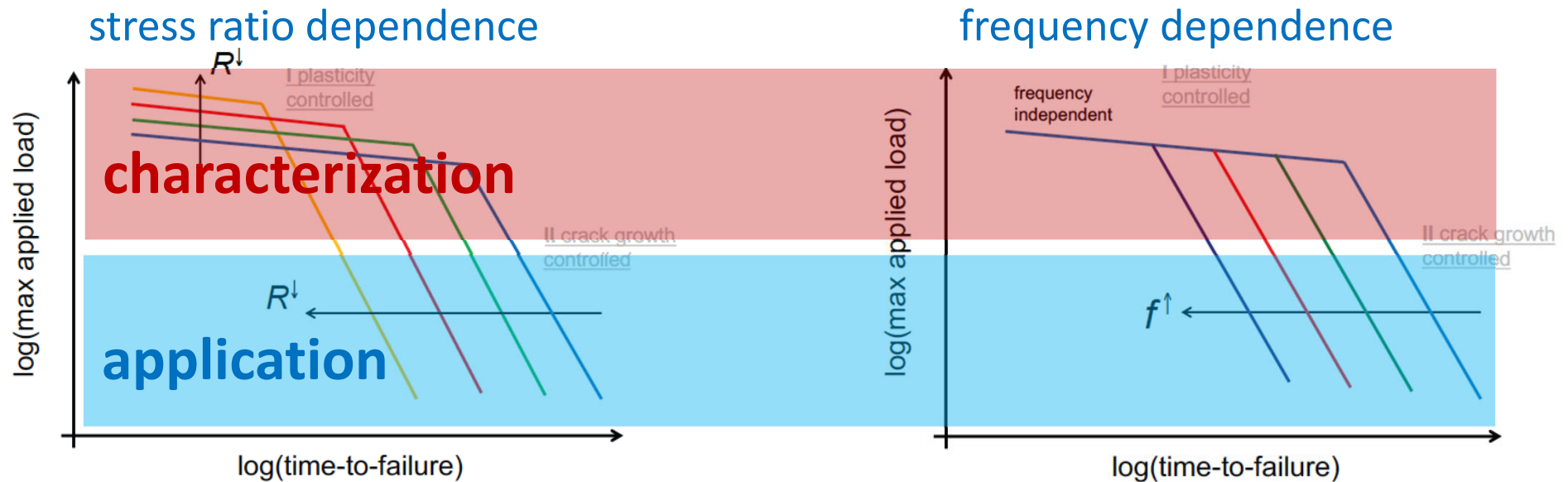
fatigue and lifetime prediction strategies for polymers



in practical applications, lifetime is likely controlled by slow crack growth

in fatigue characterization, other mechanisms may contribute !!!

fatigue and lifetime prediction strategies for polymers



correct extrapolation of “short-term” characterization data
requires identification of underlying mechanisms

Fatigue and lifetime prediction strategies for polymers

Take home messages:

- mechanical properties depend on testing conditions and molecular details
- three failure mechanism are typically identified:
 - **fracture** (Griffith's law) $=f(\text{mol.weight})$
 - **ductile flow** (Eyring's flow rule) $\neq f(\text{mol.weight})$
 - **slow crack growth** (Paris' law) $=f(\text{mol.weight})$
- under “long-term” dynamic loading conditions **ductile flow** and **slow crack growth** are the dominant failure modes:
 - **ductile flow** is a time-dependent failure mode
 - **slow crack growth** is a (strongly) frequency depend failure mode
- correct extrapolation of “short-term” characterization data requires identification of underlying mechanisms